

# ‘Stop and Drop’ Hard Lander Architectures for Europa Astrobiology Investigations.

Kevin P. Hand

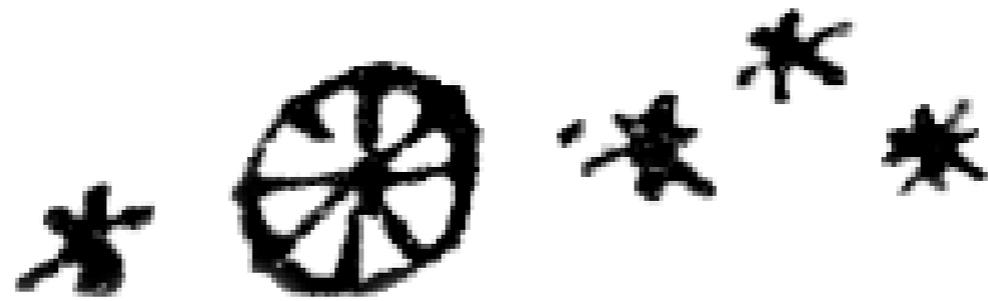
A. Sengstacken, M. Rudolph, J. Lang, R. Amini, J. Ludwinski  
Jet Propulsion Laboratory,  
California Institute of Technology

IKI Europa Lander Conference, Feb. 9-13, Moscow, Russia

Vladimir Krasnopolsky, Lev Mukhin, Vasily Moroz, Kerzhanovich, 1985



Image courtesy of D. Cruikshank



# Kuiper 1957 AAS Meeting

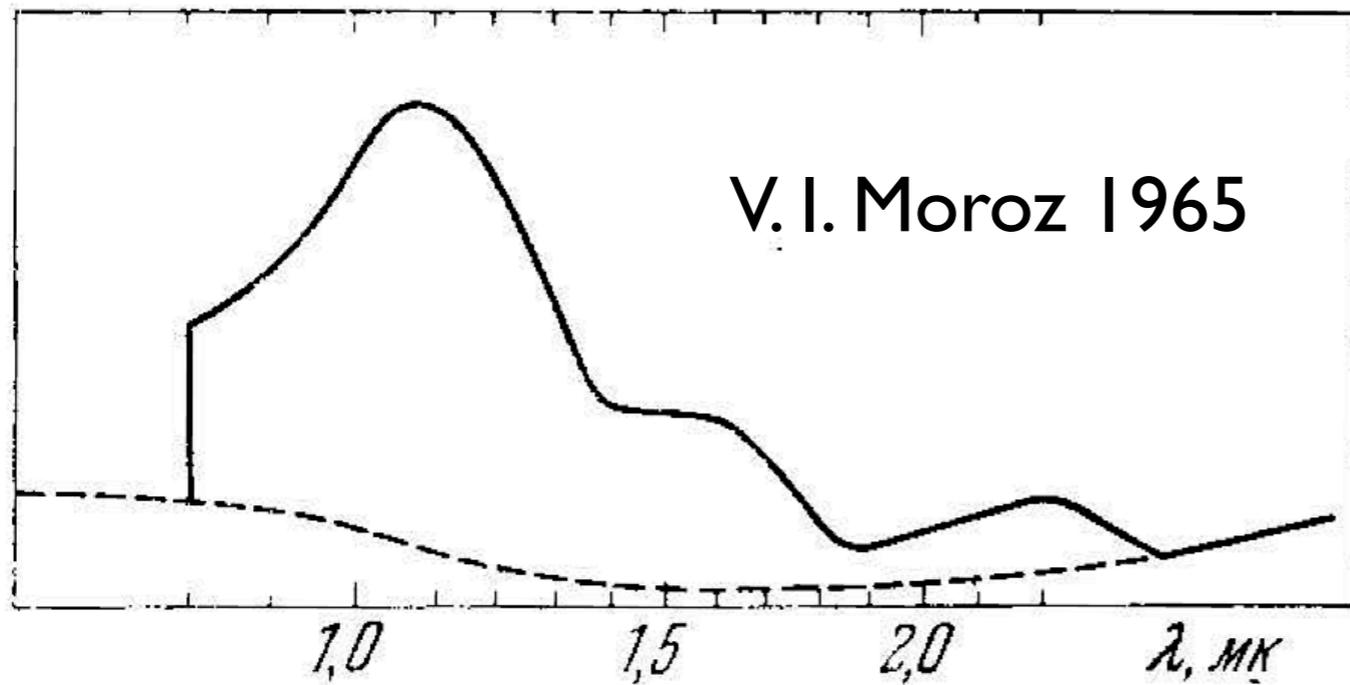
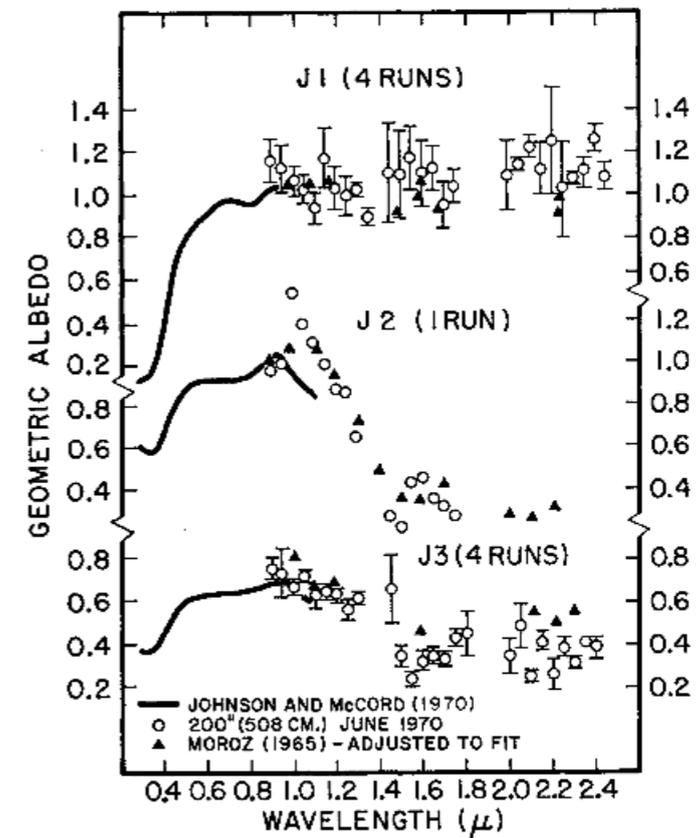
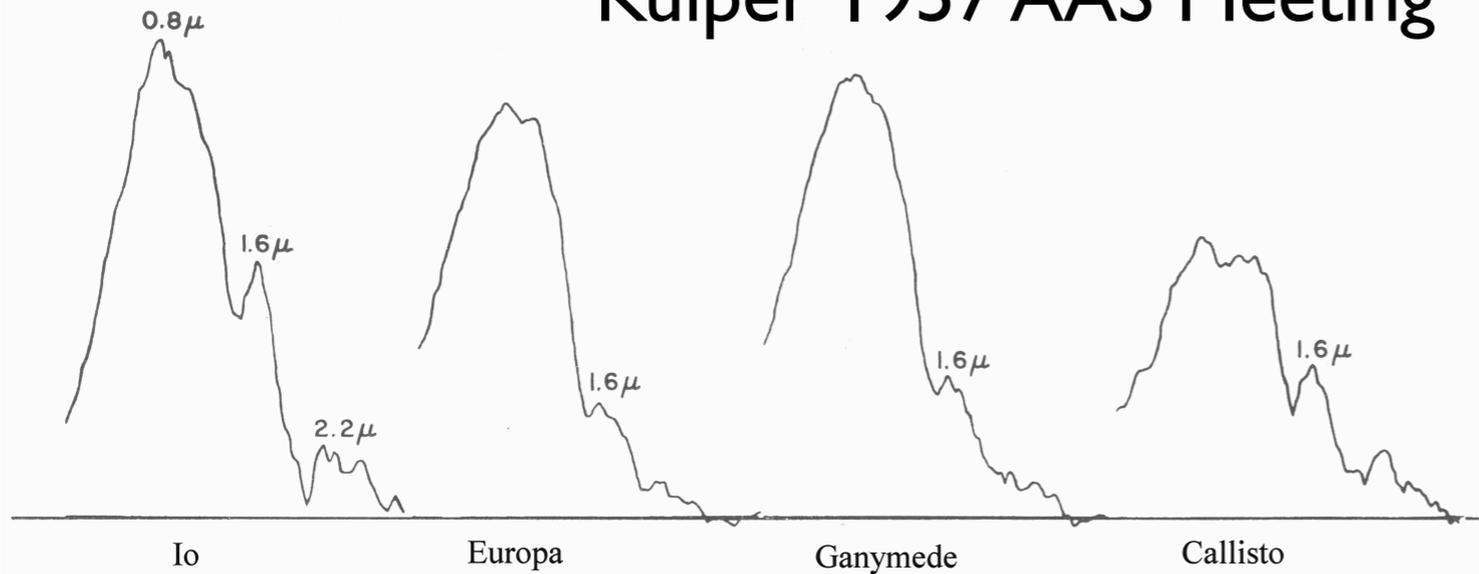
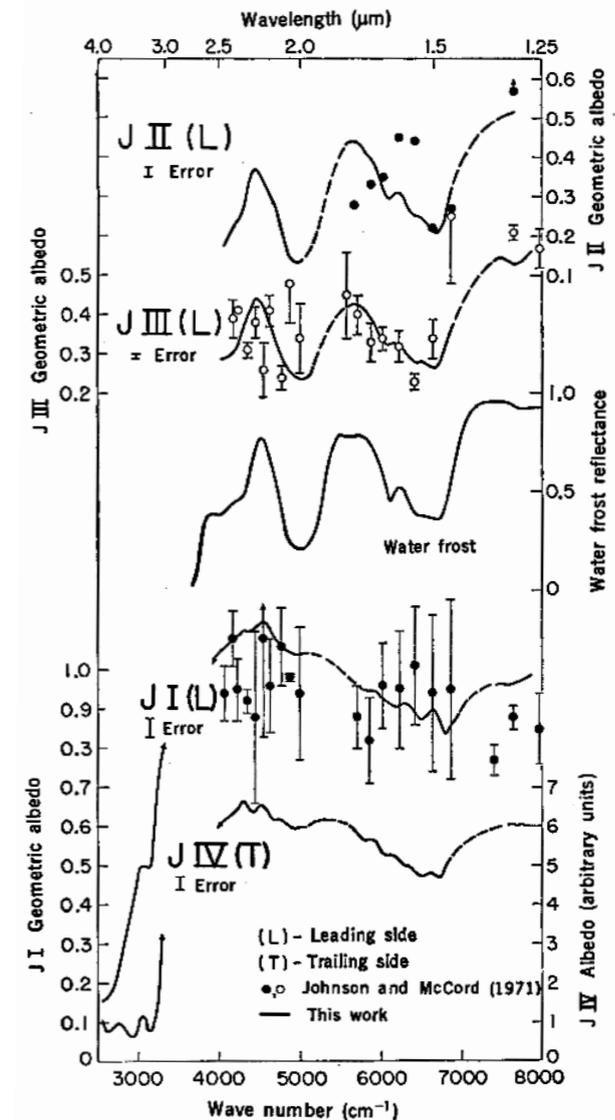
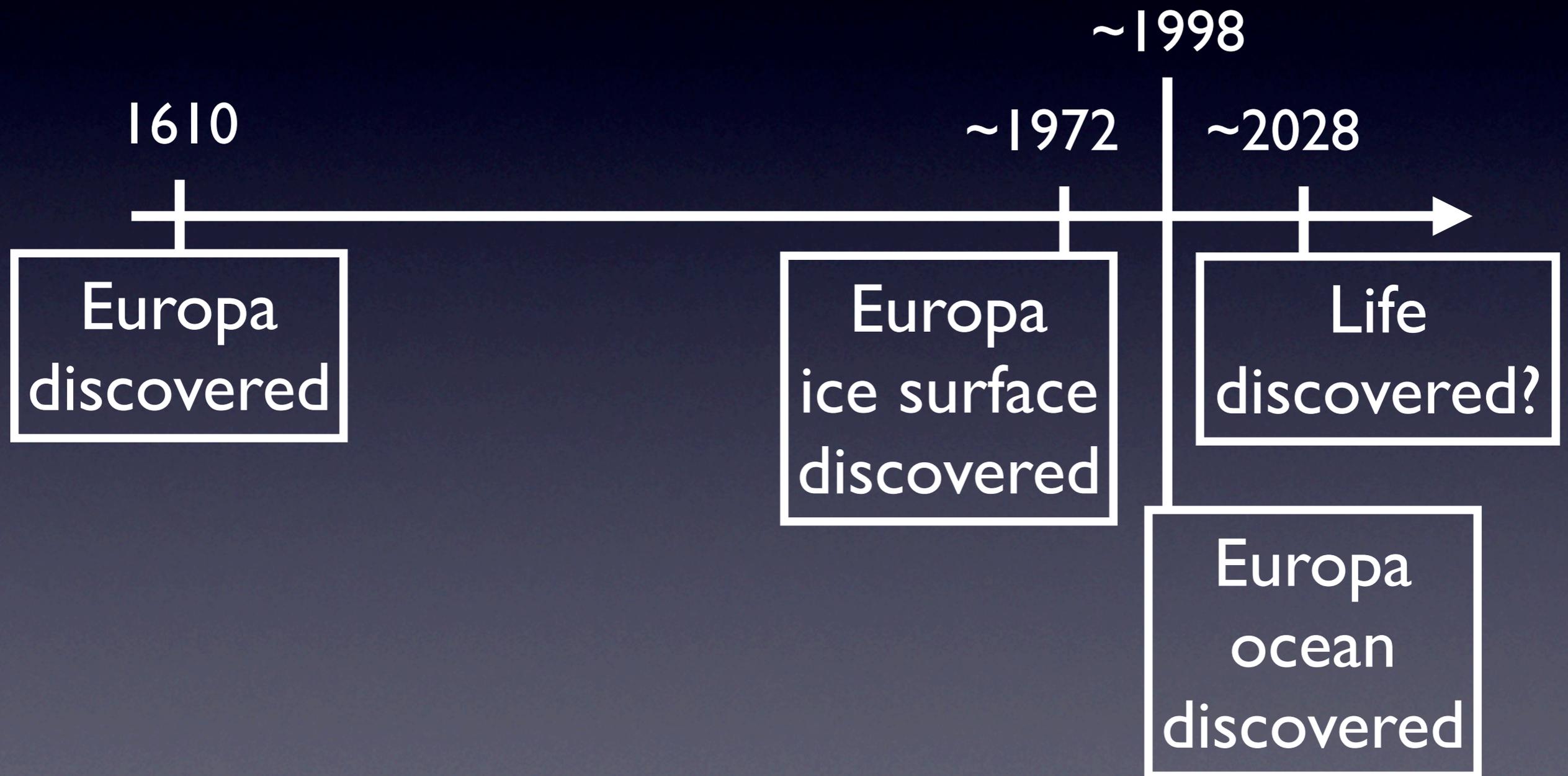


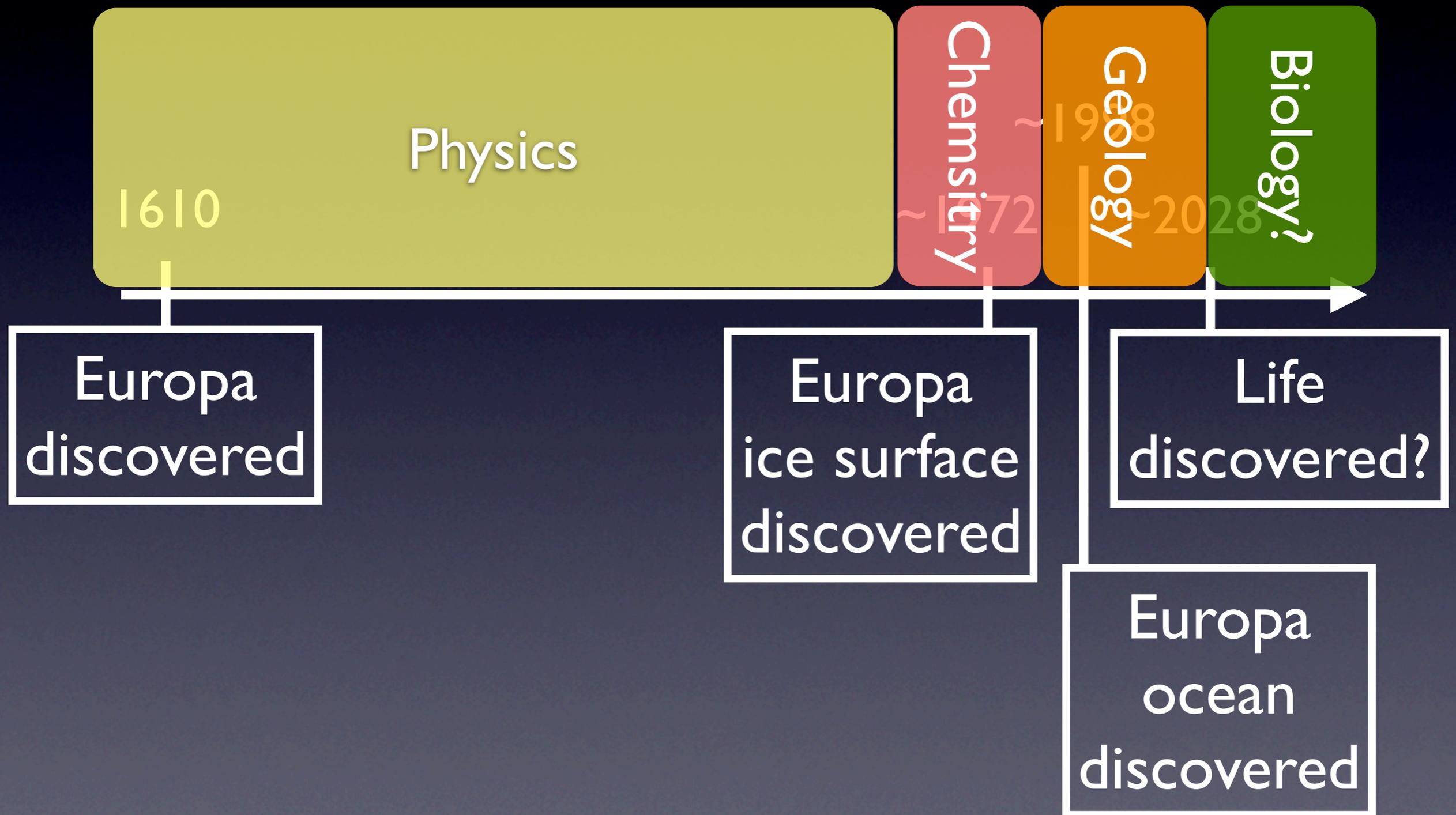
Рис. 198. Спектр Европы, среднее из четырех записей 1.10 1964 г., ЗТШ, Нуль-пункт (пунктир) зависит от длины волны вследствие слабой паразитной подсветки.



# Europa Astrobiology



# Europa Astrobiology

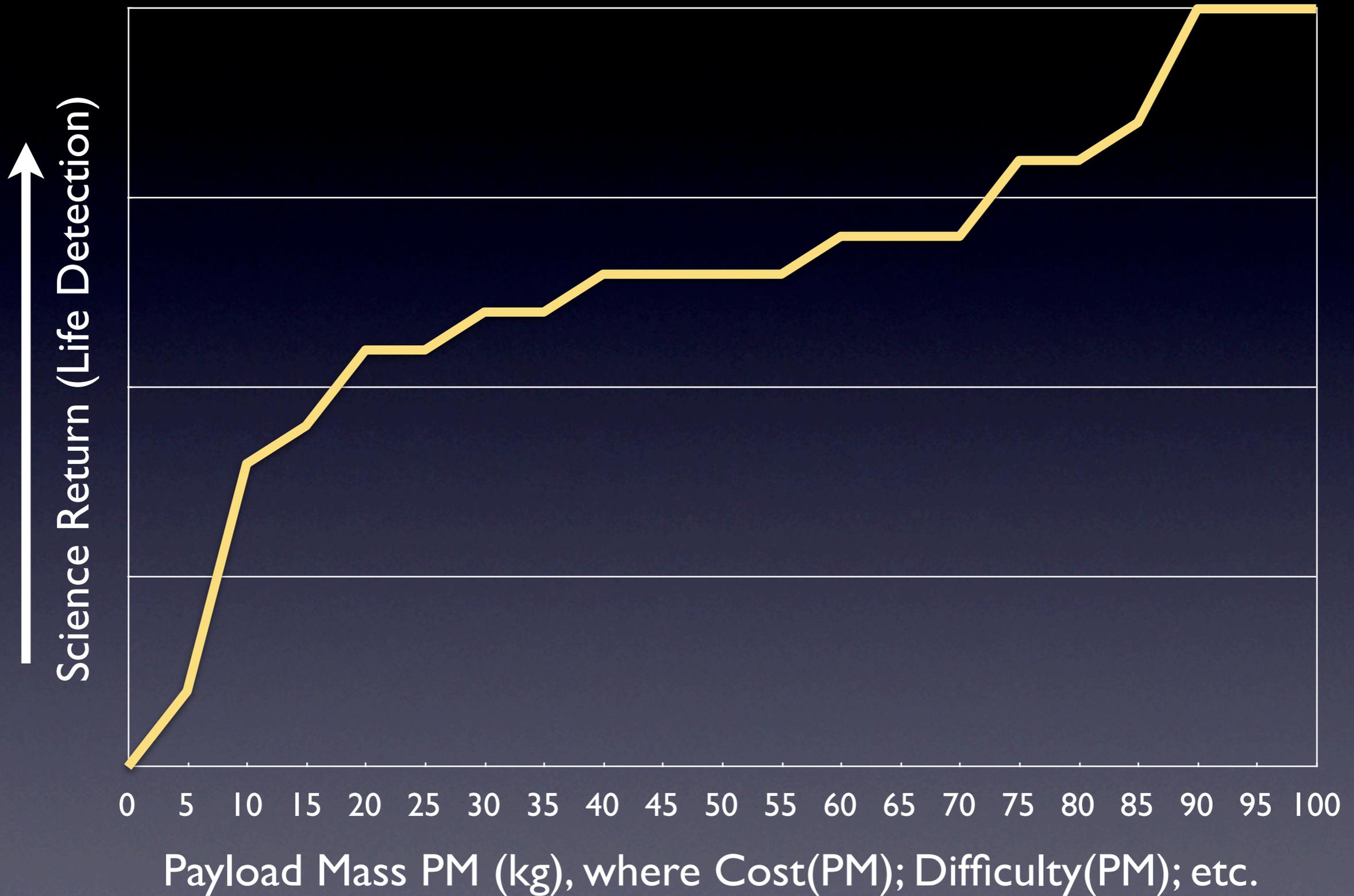


# Clark et al. Europa Orbiter 2008 Studies

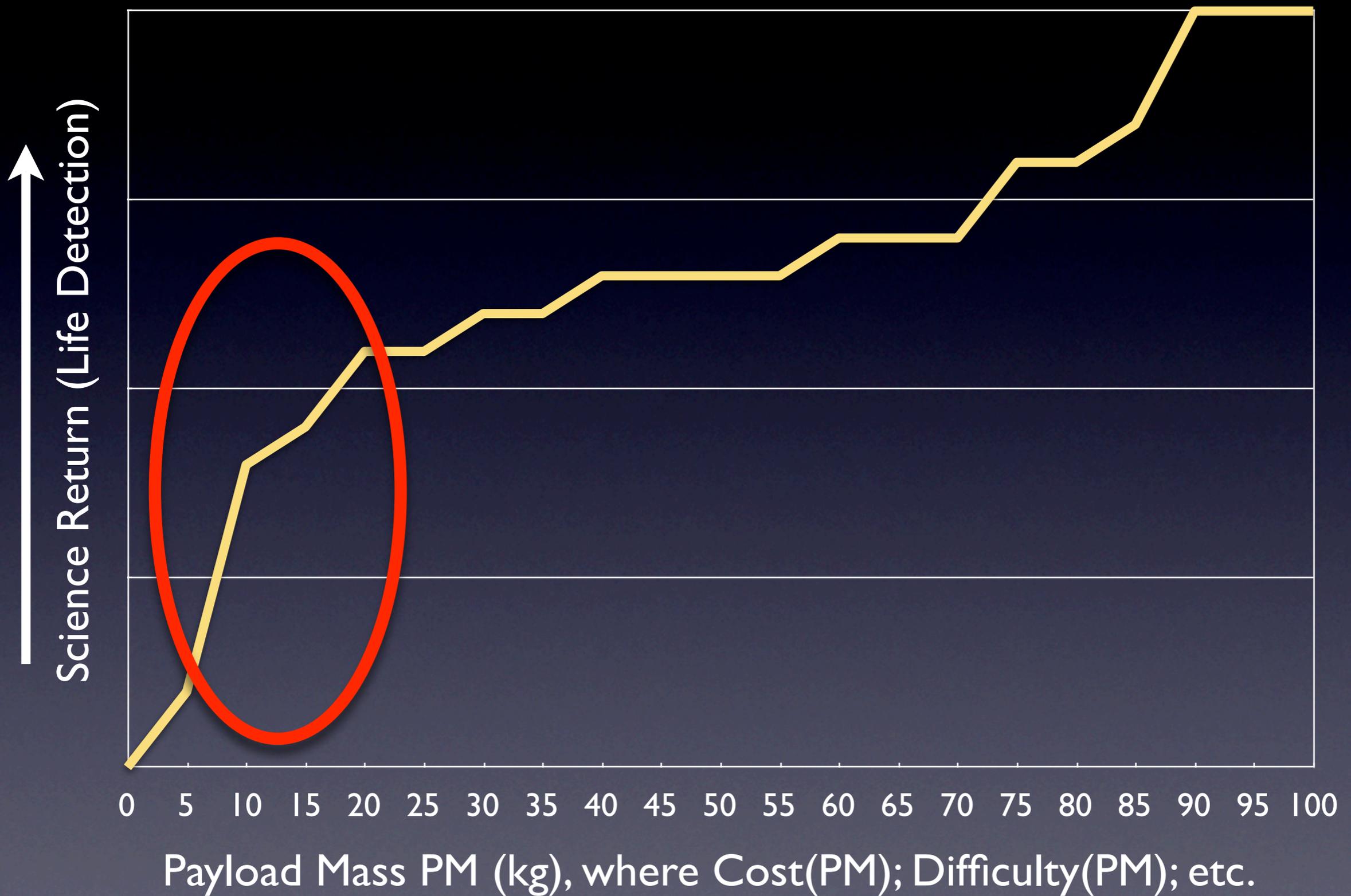
Atlas V 551 Launch Capability	
Launch	Mass margin (kg)
Oct. 2018	130
June 2019	241
Dec. 2019	-116
March 2020	262
March 2020	327
May 2021	73
Nov. 2021	165
March 2023	-40

Mass available after adding 'sweet spot' instruments. Does not account for accommodation, margin, shielding, or power needs.

# Science Return as a function of instrumentation delivered to surface



# Science Return as a function of instrumentation delivered to surface



# Relevance to recommendations and science goals of the NASA Decadal Survey\*

Goal 2.1: Characterize the surface composition, especially compounds of interest to prebiotic chemistry.

Theme 2.1: What is the chemical composition of the water-rich phase?

Theme 2.4: Can and does life exist in the internal ocean of an icy satellite?

Theme 4.2.1: Is there extant life in the outer solar system?

\*Goals overlap significantly with the NASA Astrobiology Roadmap (2008).

# Relevance to recommendations and science goals of the NASA Decadal Survey\*

Goal 2.1: Characterize the surface composition, especially compounds of interest to prebiotic

## Critical advantages of *in situ* analysis

- Ground truth orbital measurements
- Enhanced concentration/Detection limits

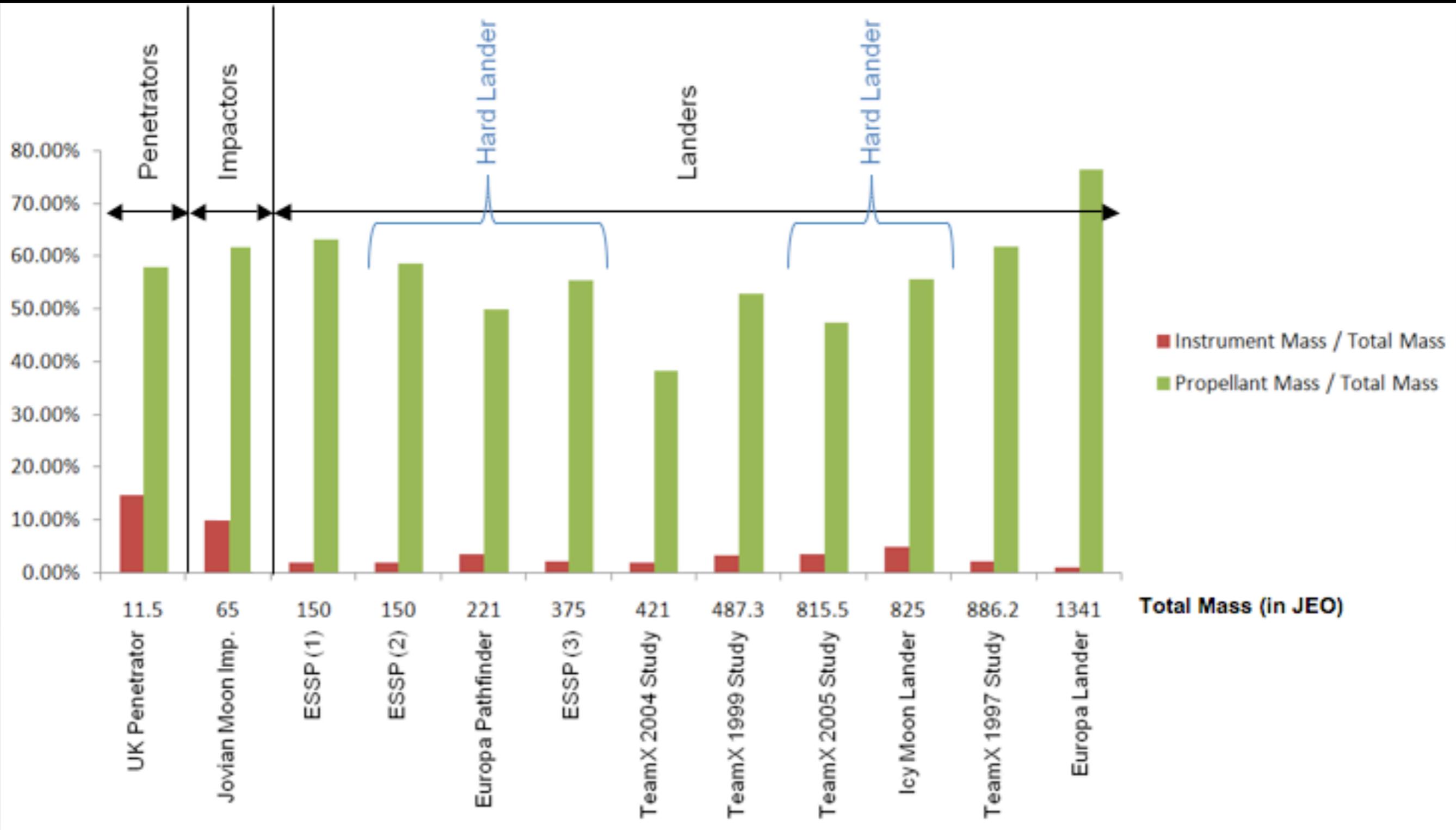
Theme 4.2.1: Is there extant life in the outer solar system?

\*Goals overlap significantly with the NASA Astrobiology Roadmap (2008).

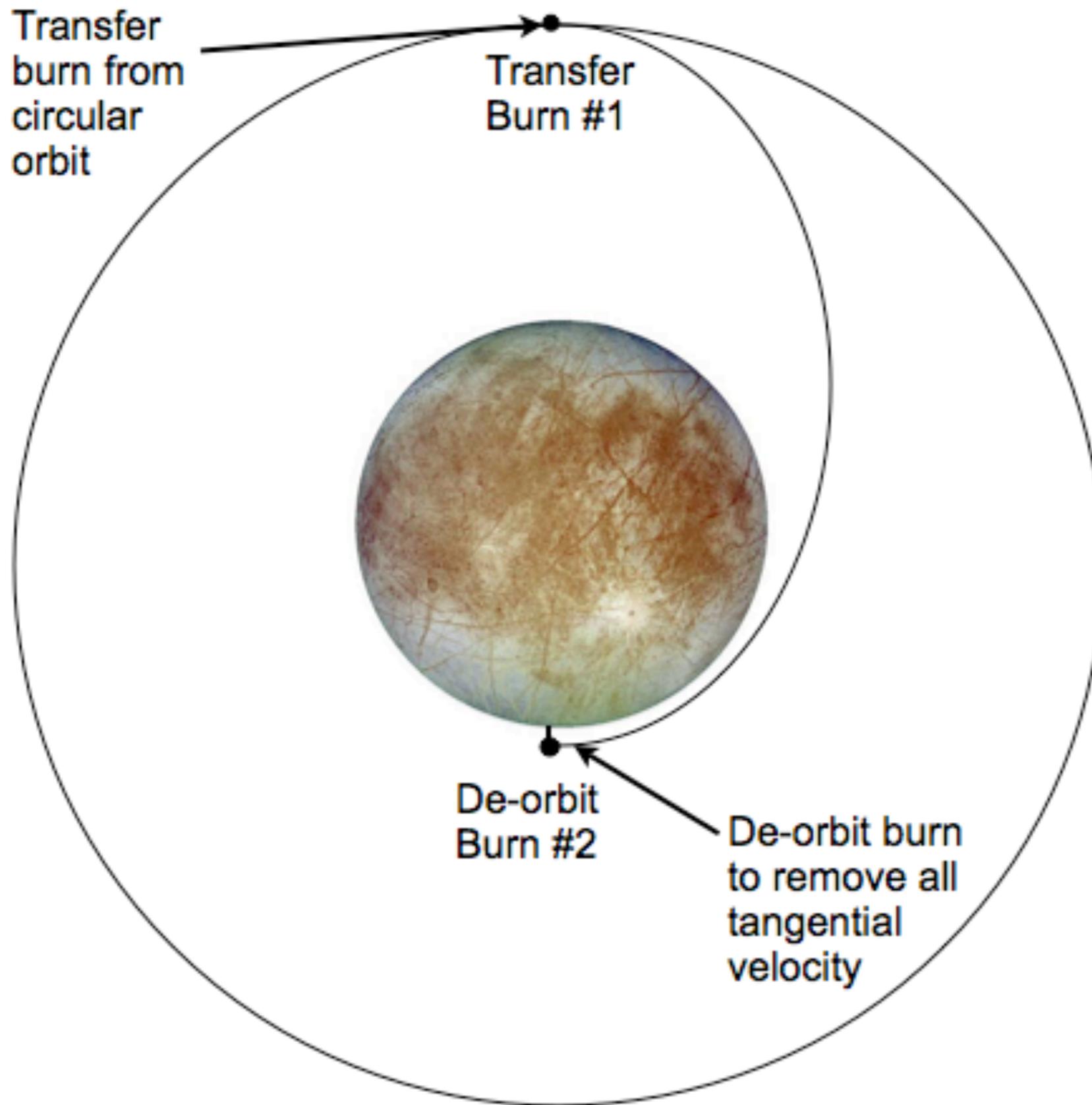
# Definitions

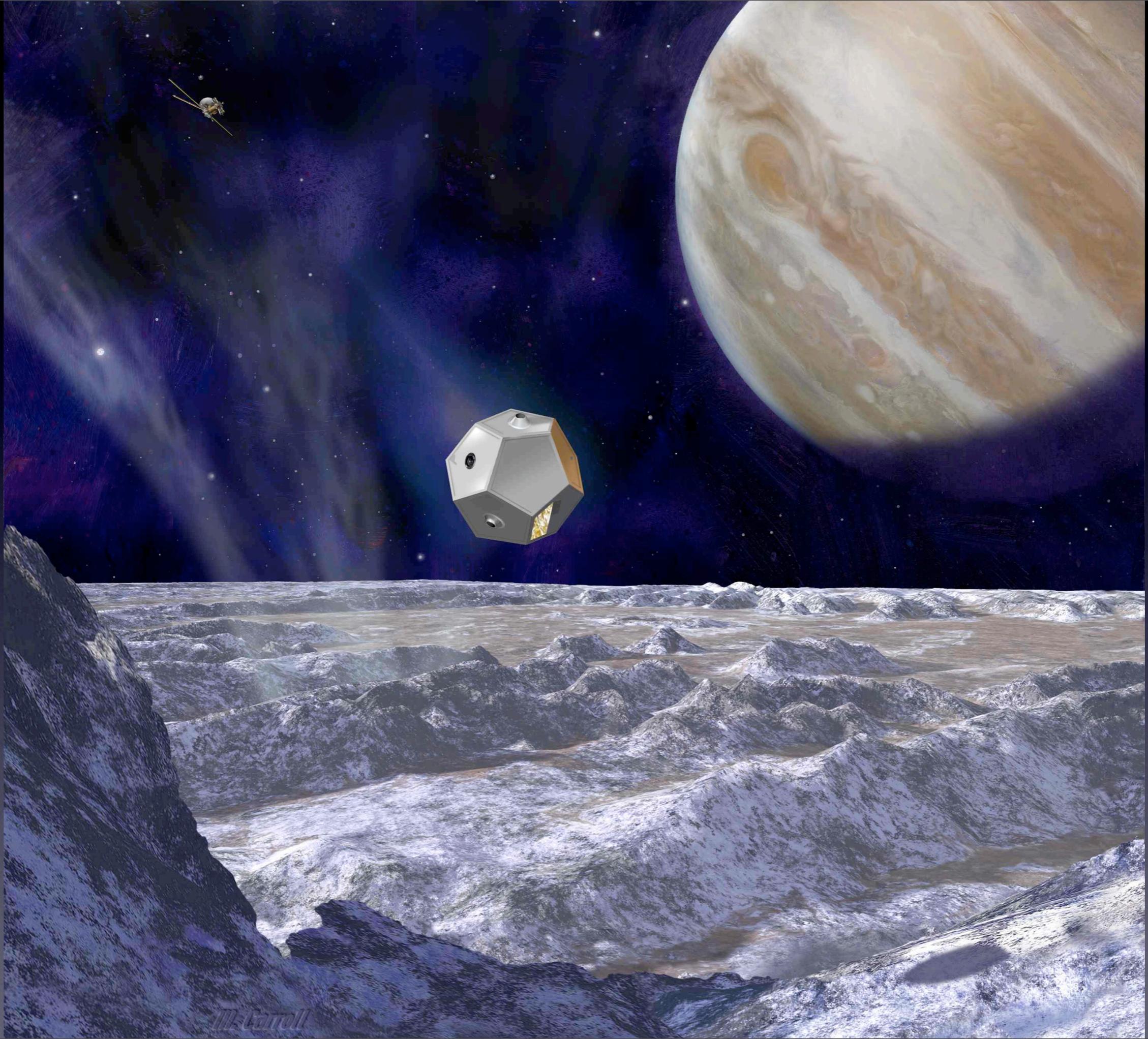
Sub Satellite	Penetrator	Impactor
<ul style="list-style-type: none"> <li>Performs science in orbit or during descent to mission termination impact</li> </ul>	<ul style="list-style-type: none"> <li>Performs sub-surface science</li> <li>Impacts at high velocity, penetrating the surface like a bullet</li> </ul>	<ul style="list-style-type: none"> <li>Performs science during descent</li> <li>Impact with the surface is designed to shoot ejecta up to be analyzed from orbit</li> </ul>

Lander		
Performs science on the surface		
Very Hard	Hard	Soft
<ul style="list-style-type: none"> <li>Drops from as high as 10 km</li> </ul>	<ul style="list-style-type: none"> <li>Drops from as high as 1 km or as low as 10m</li> </ul>	<ul style="list-style-type: none"> <li>Touches down with minimal accelerations</li> </ul>



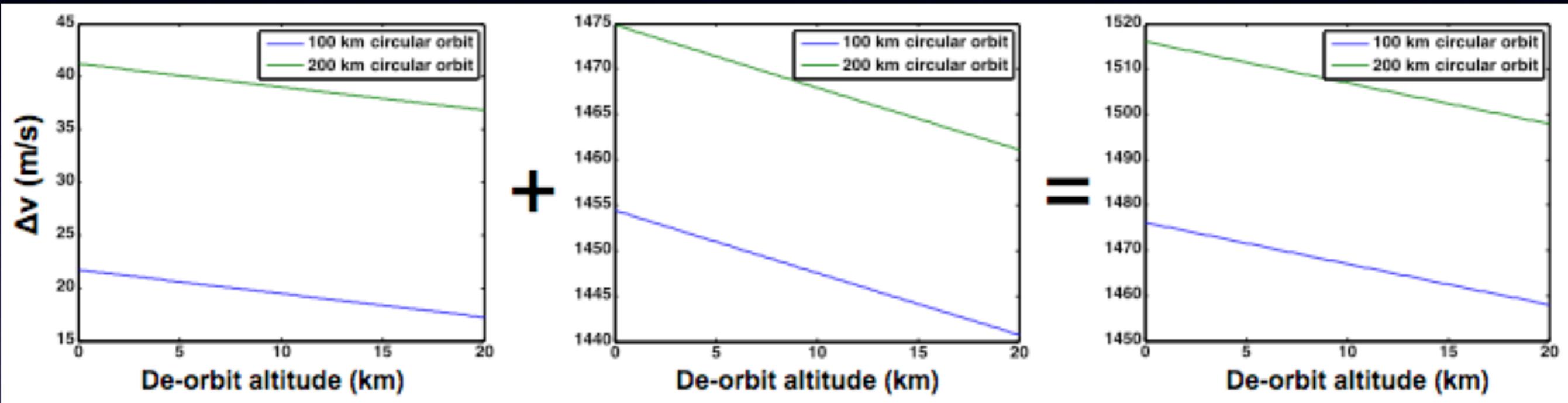
# “Stop and Drop” Hard Lander





M. Carroll

# Burn $\Delta v$ 's and deorbit altitude

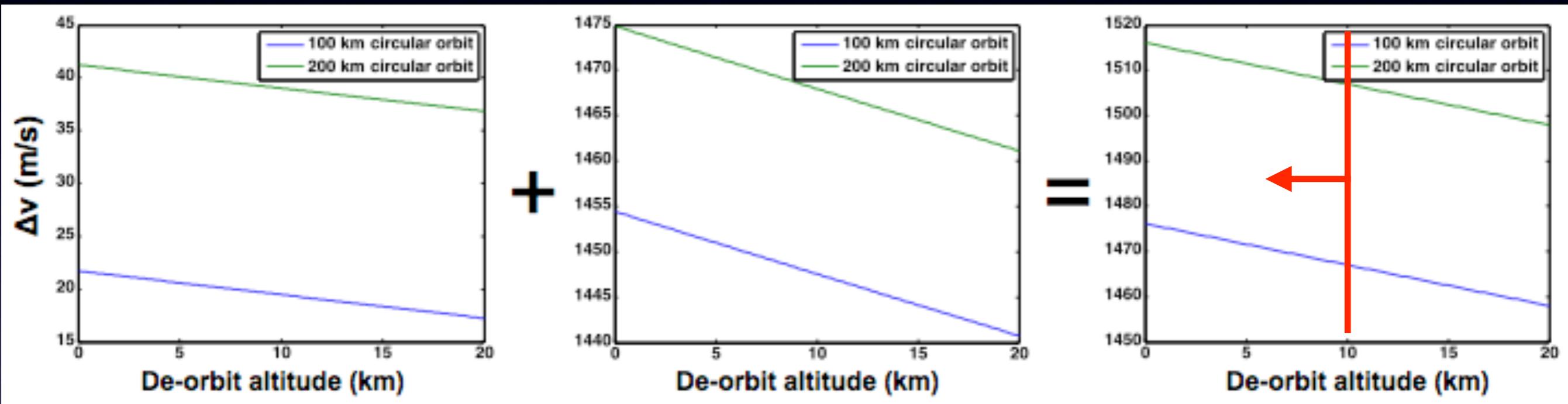


Burn 1

Burn 2

'Drop Point'

# Burn $\Delta v$ 's and deorbit altitude

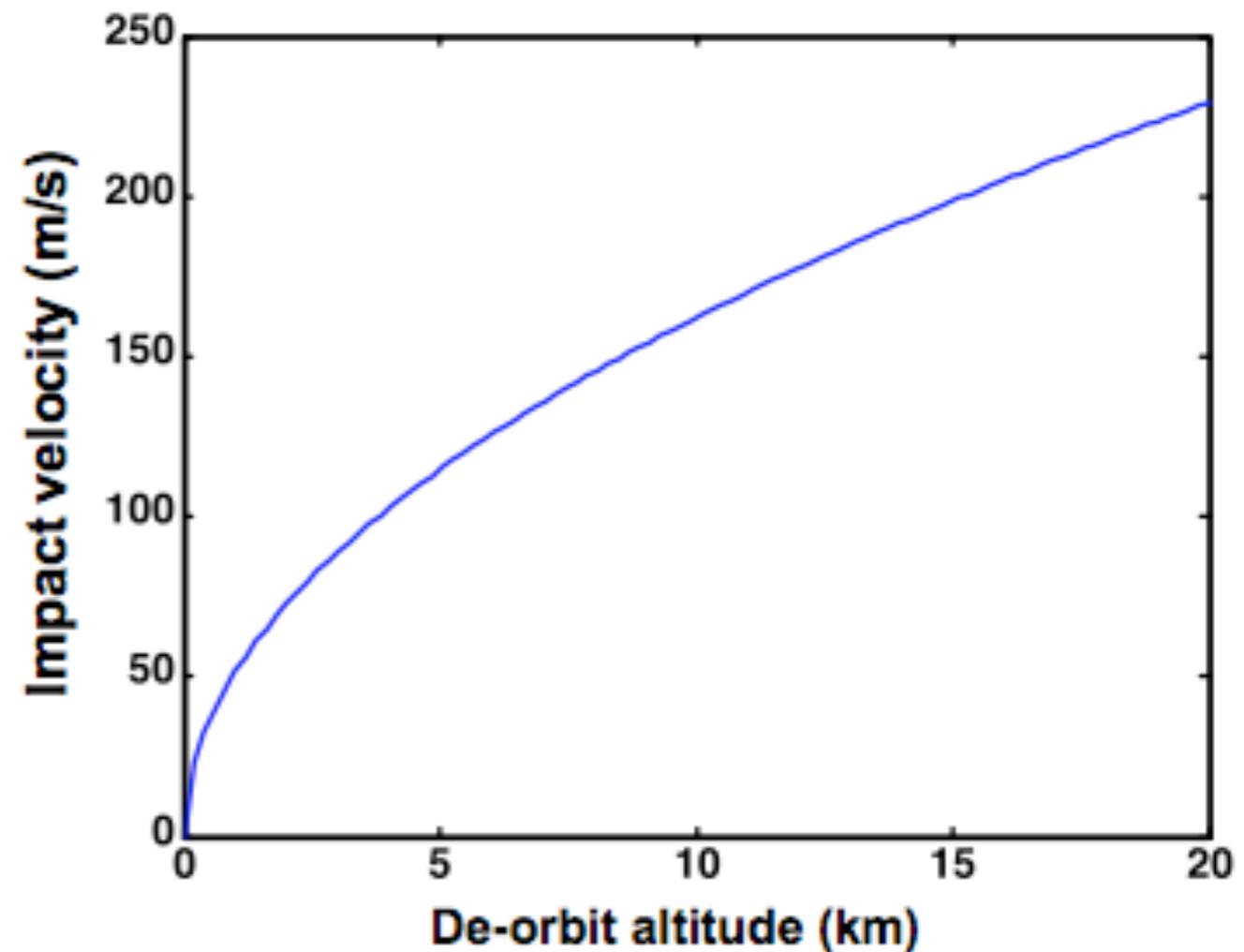
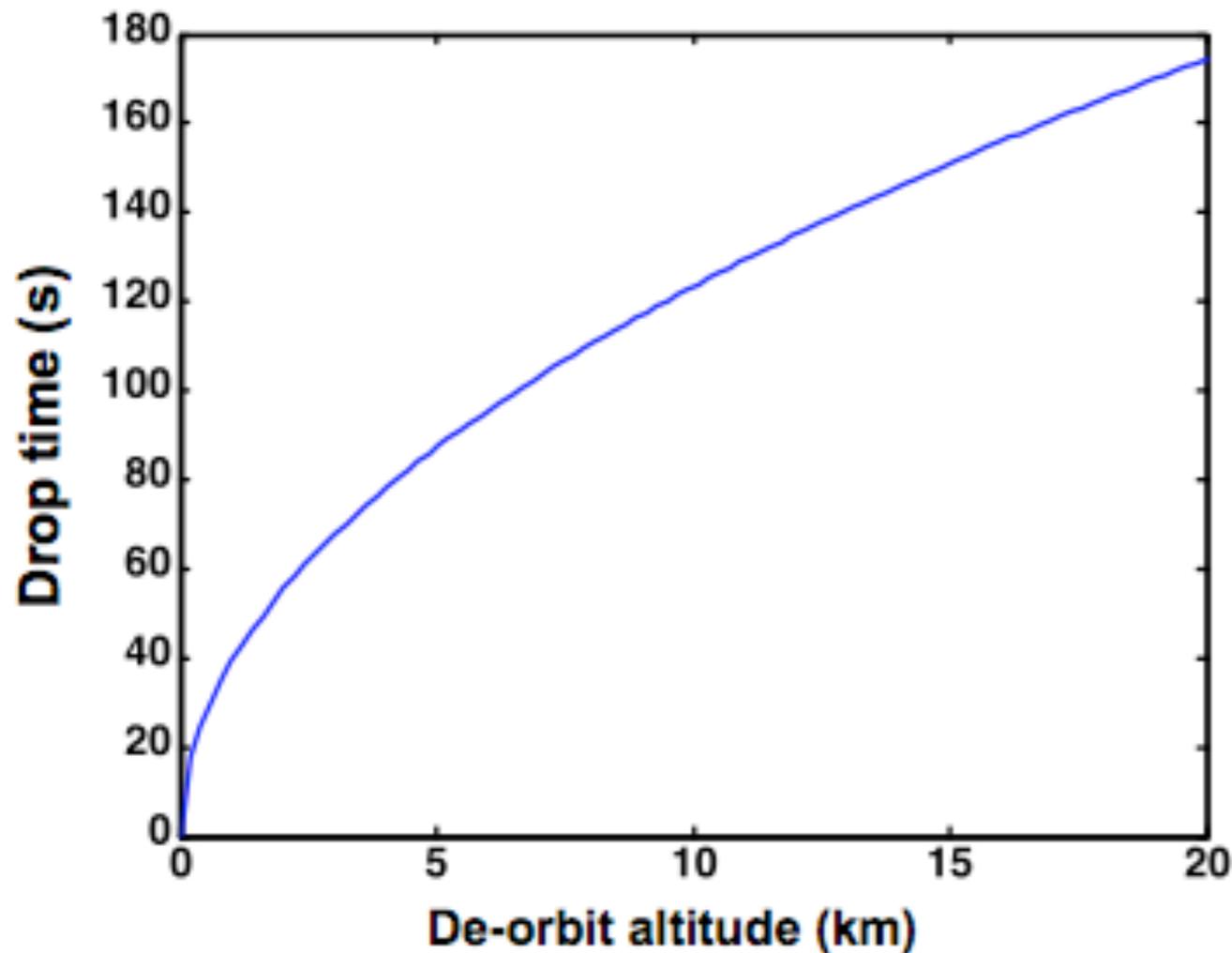


Burn 1

Burn 2

'Drop Point'

# Impact Velocity & Drop Time

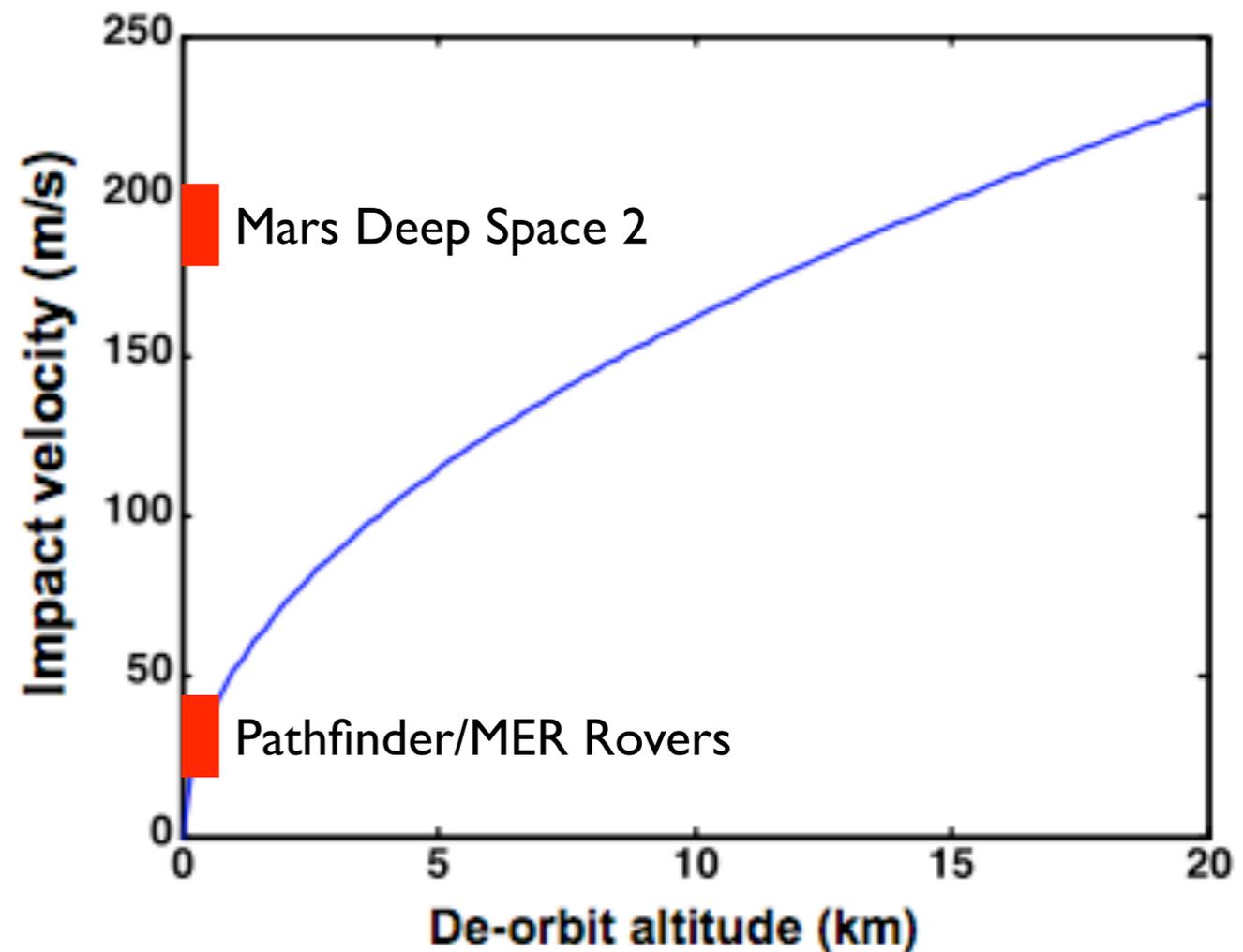
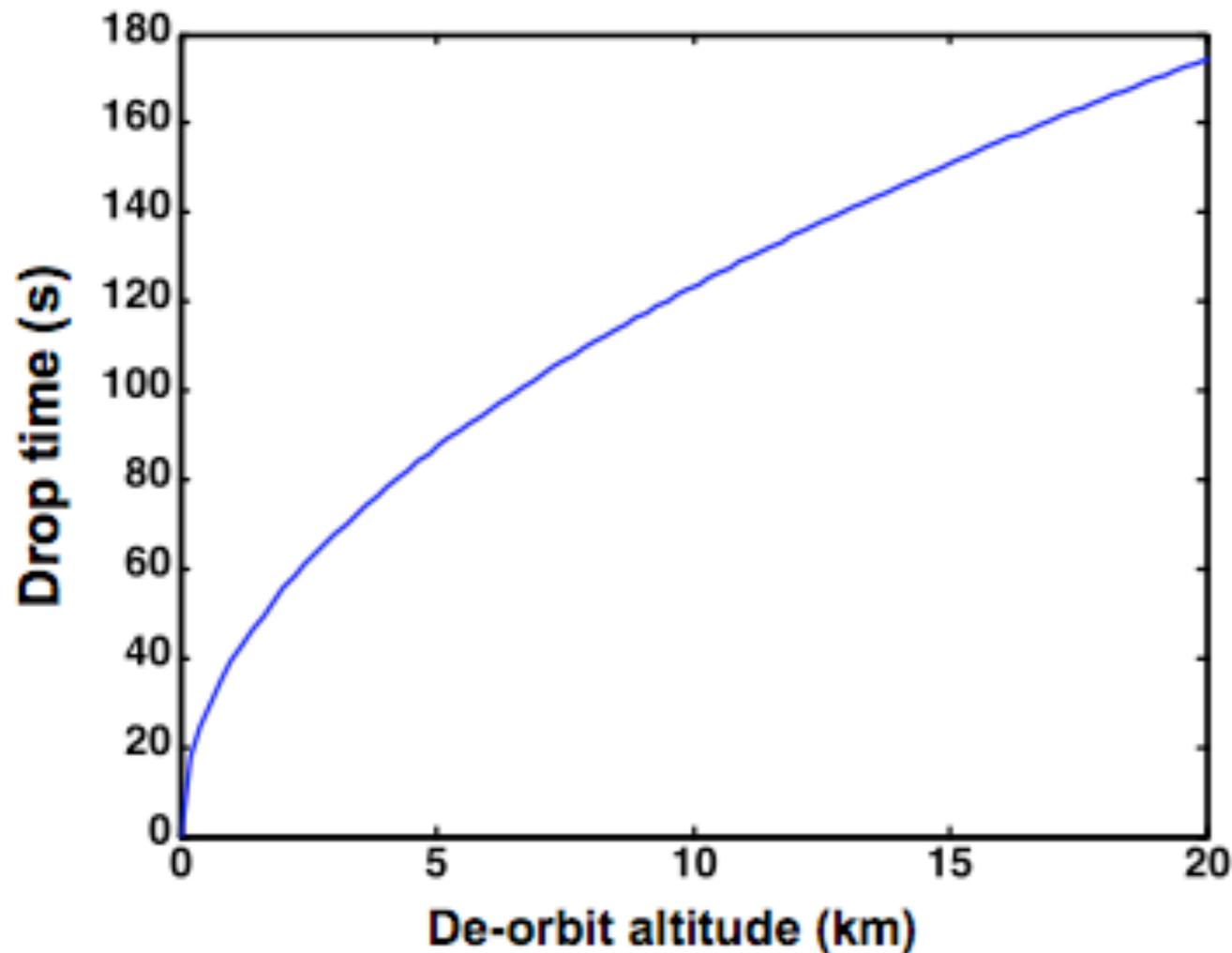


200 km orbit  $\Delta v \sim 1510$  km/s

100 km orbit  $\Delta v \sim 1465$  km/s

100 kg probe: 35-55 kg propellant and Isp 200-360 s

# Impact Velocity & Drop Time

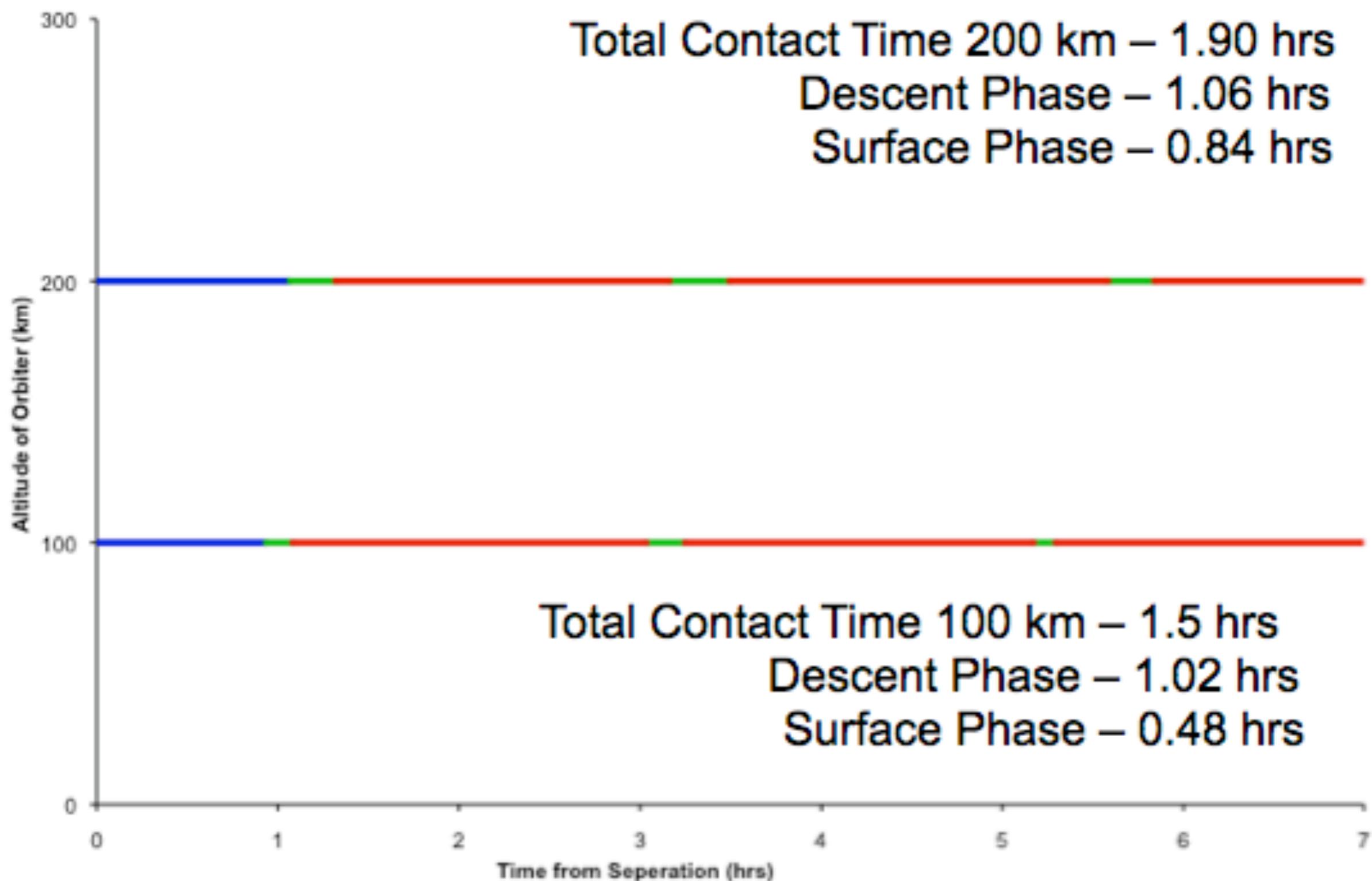


200 km orbit  $\Delta v \sim 1510$  km/s

100 km orbit  $\Delta v \sim 1465$  km/s

100 kg probe: 35-55 kg propellant and Isp 200-360 s

Orbiter to Lander    Orbiter to Ground    No Contact

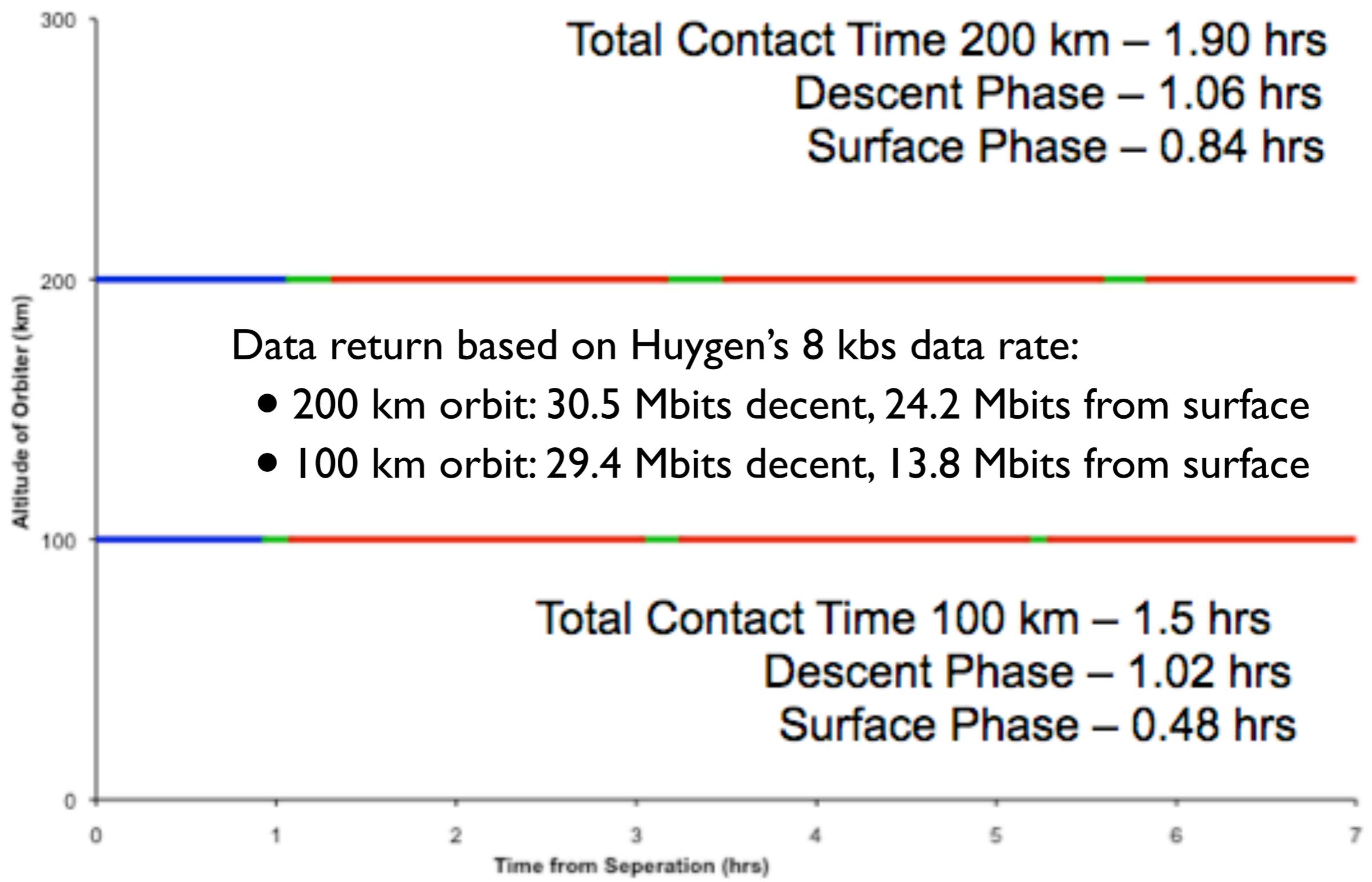


**Total Contact Time 200 km – 1.90 hrs**  
**Descent Phase – 1.06 hrs**  
**Surface Phase – 0.84 hrs**

**Total Contact Time 100 km – 1.5 hrs**  
**Descent Phase – 1.02 hrs**  
**Surface Phase – 0.48 hrs**



**Total Contact Time 200 km – 1.90 hrs**  
**Descent Phase – 1.06 hrs**  
**Surface Phase – 0.84 hrs**



Data return based on Huygen's 8 kbs data rate:

- 200 km orbit: 30.5 Mbits decent, 24.2 Mbits from surface
- 100 km orbit: 29.4 Mbits decent, 13.8 Mbits from surface

**Total Contact Time 100 km – 1.5 hrs**  
**Descent Phase – 1.02 hrs**  
**Surface Phase – 0.48 hrs**

# Lander Lifetime

(Science, Radiation & Power Considerations)

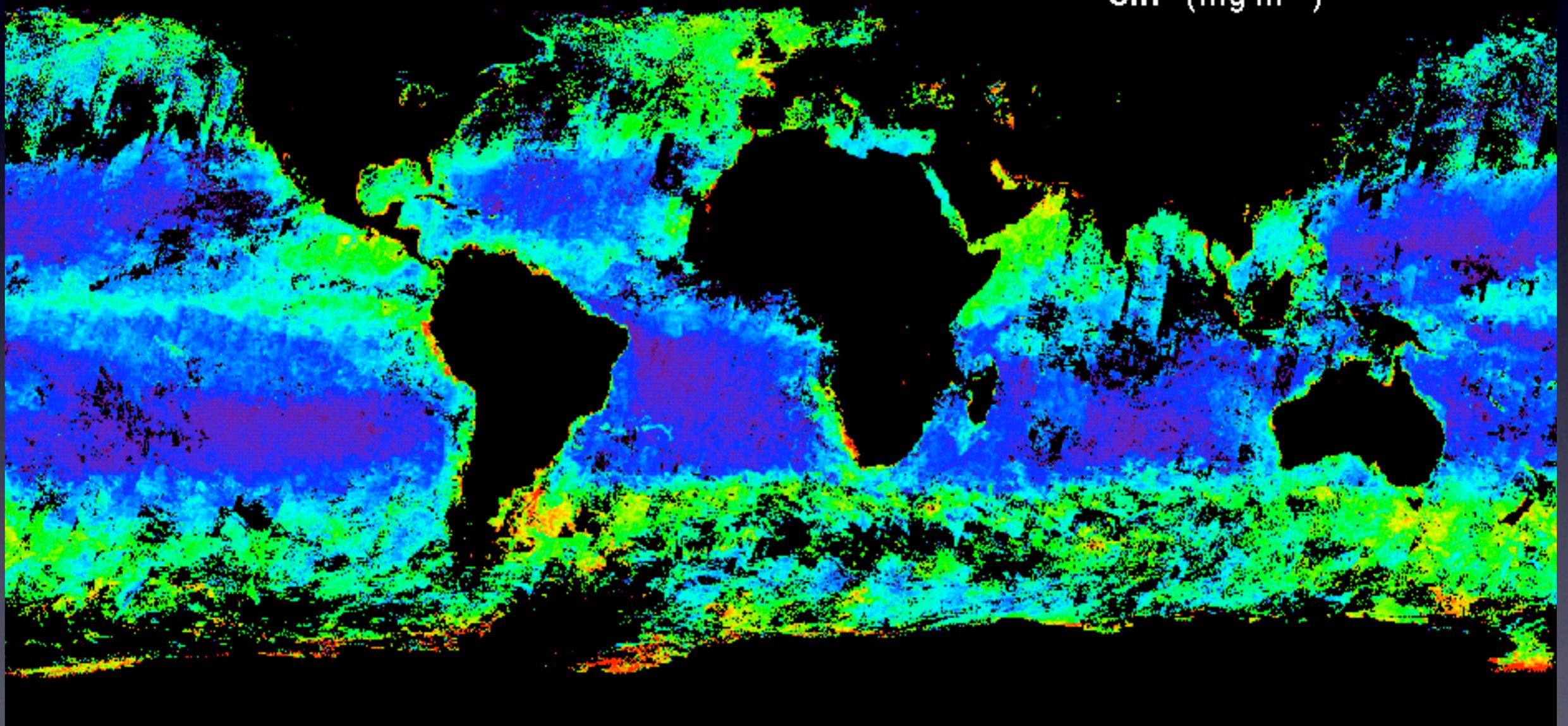
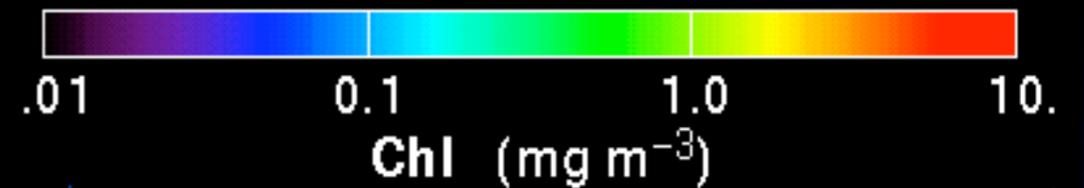
<p>~7 hours</p>	<p>First phase: decent plus 3 orbits</p>
<p>~30 hours</p>	<p>First phase, plus 8.5 hour 'dark' phase, then 13 hrs for next 6-7 orbits</p>
<p>~7 + <math>n21.5</math> hours</p>	<p>First phase, plus 21.5 hours to achieve each increment of 'dark' phase plus subsequent communication orbits.</p>
<p>Full European day (85.2 hours)</p>	<p>Lander should survive for ~93 hours in order to communicate all data back to orbiter.</p>

# Lessons learned from Viking Landers

- 1) If the payload permits, conduct experiments that assume contrasting definitions for life.
- 2) Given limited payload, the biochemical definition deserves priority.
- 3) Establishing the geological and chemical context of the environment is critical.
- 4) Life-detection experiments should provide valuable information regardless of the biology results.
- 5) Exploration need not, and often cannot, be hypothesis testing. Planetary missions are often missions of exploration, and therefore the above guidelines must be put in the context of exploration and discovery driven science.

# Chlorophyll-a mapping of Earth's ocean

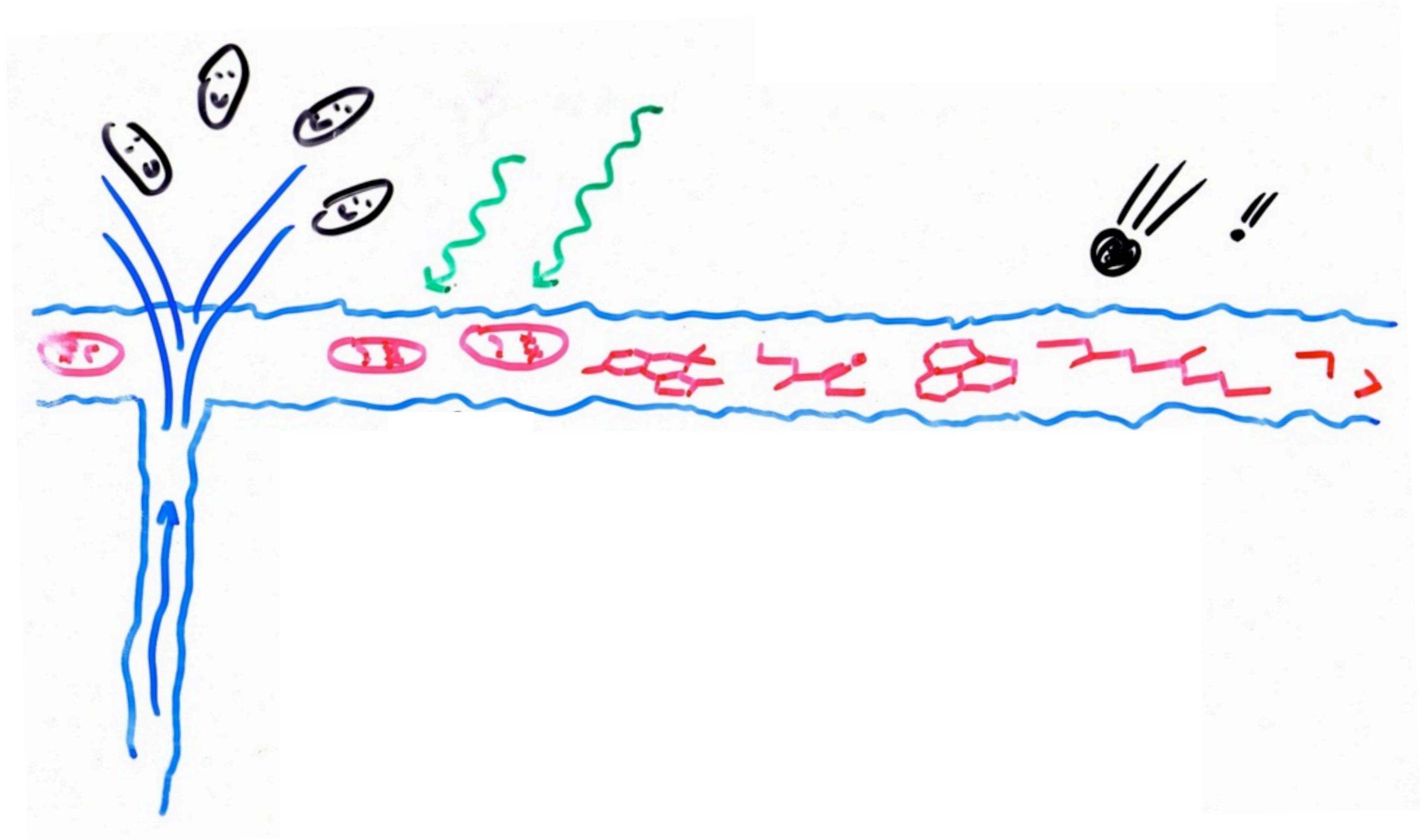
Jan 25, 2001

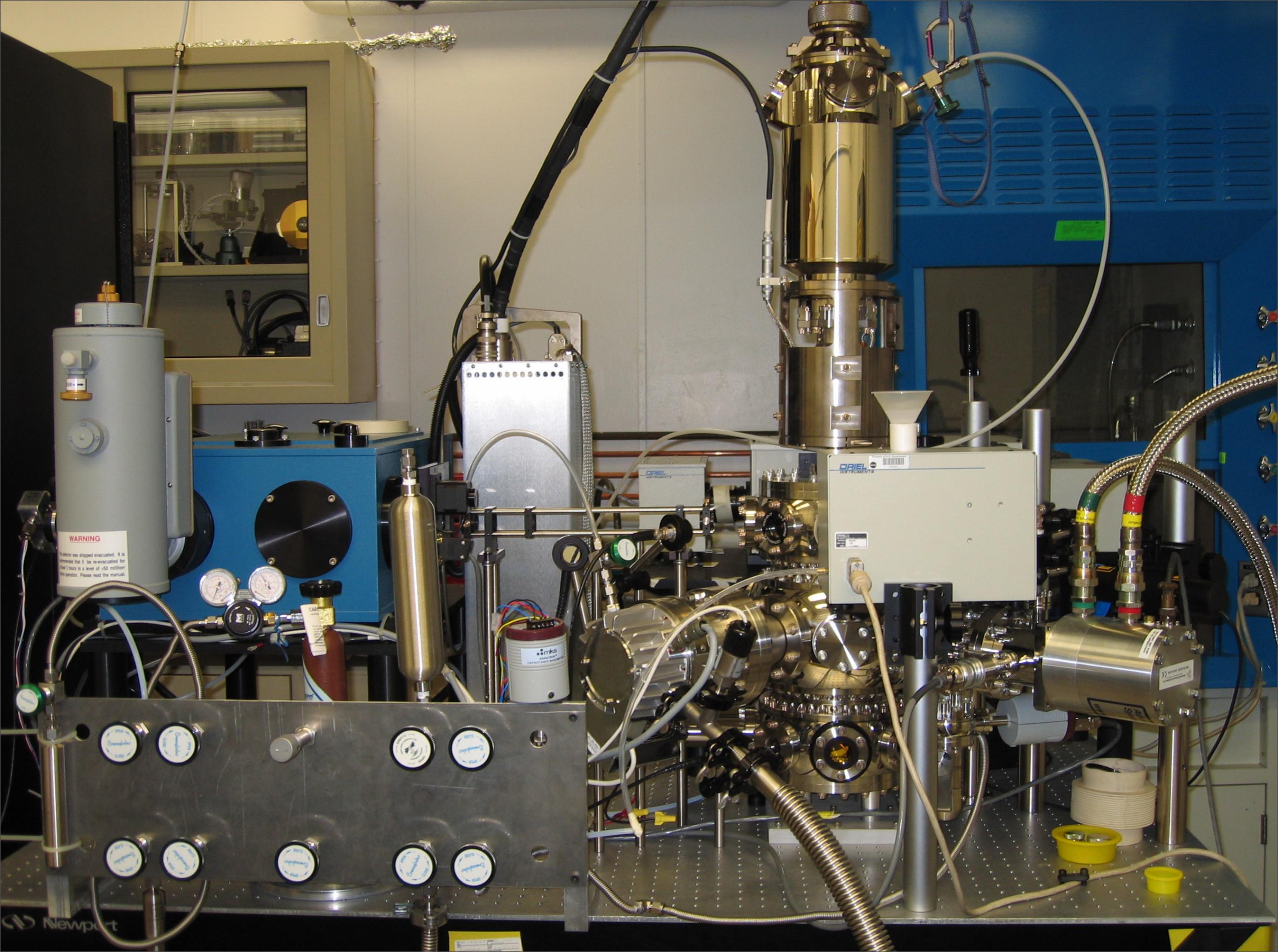


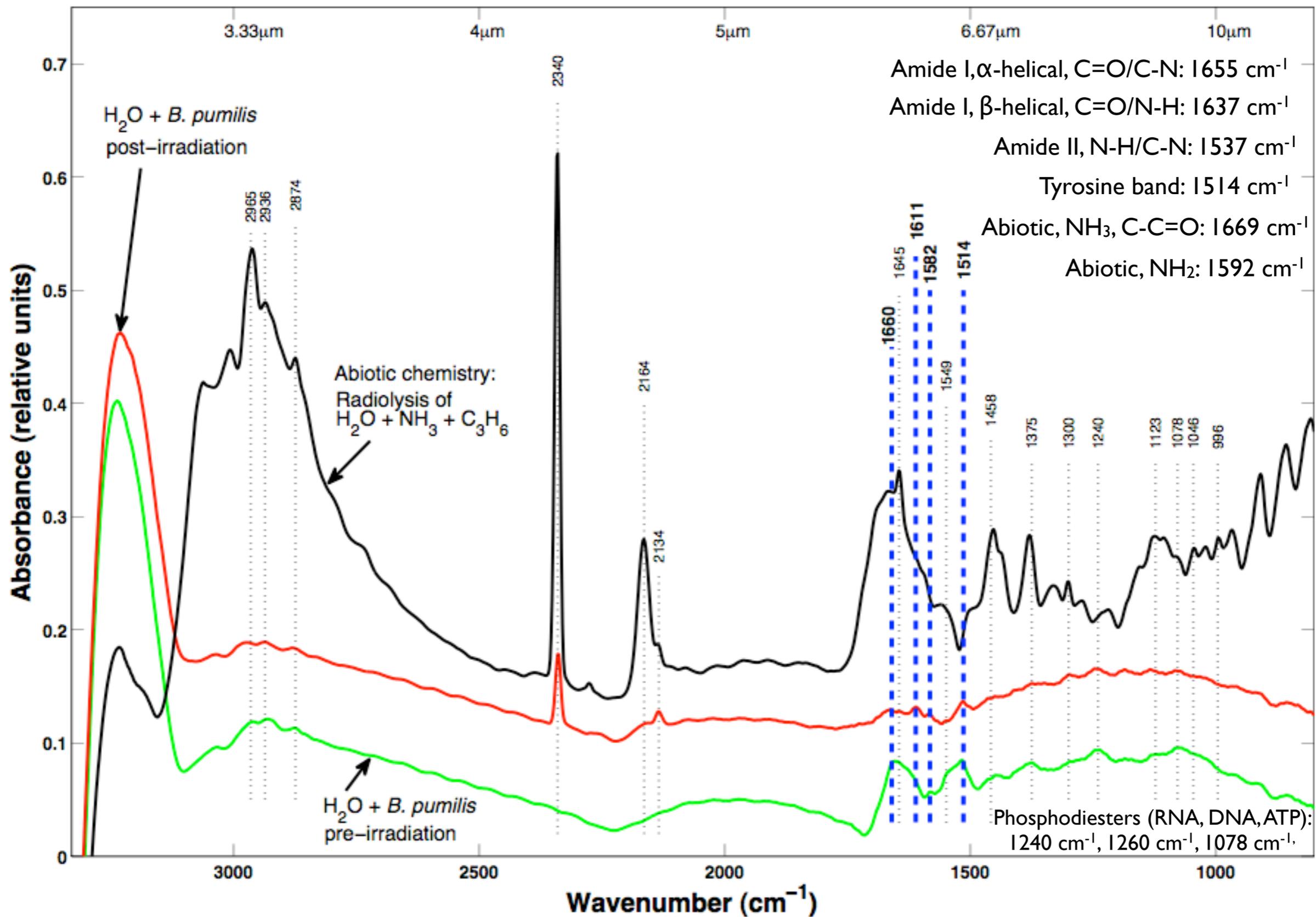
Terra satellite, Moderate Resolution Imaging Spectroradiometer (MODIS)

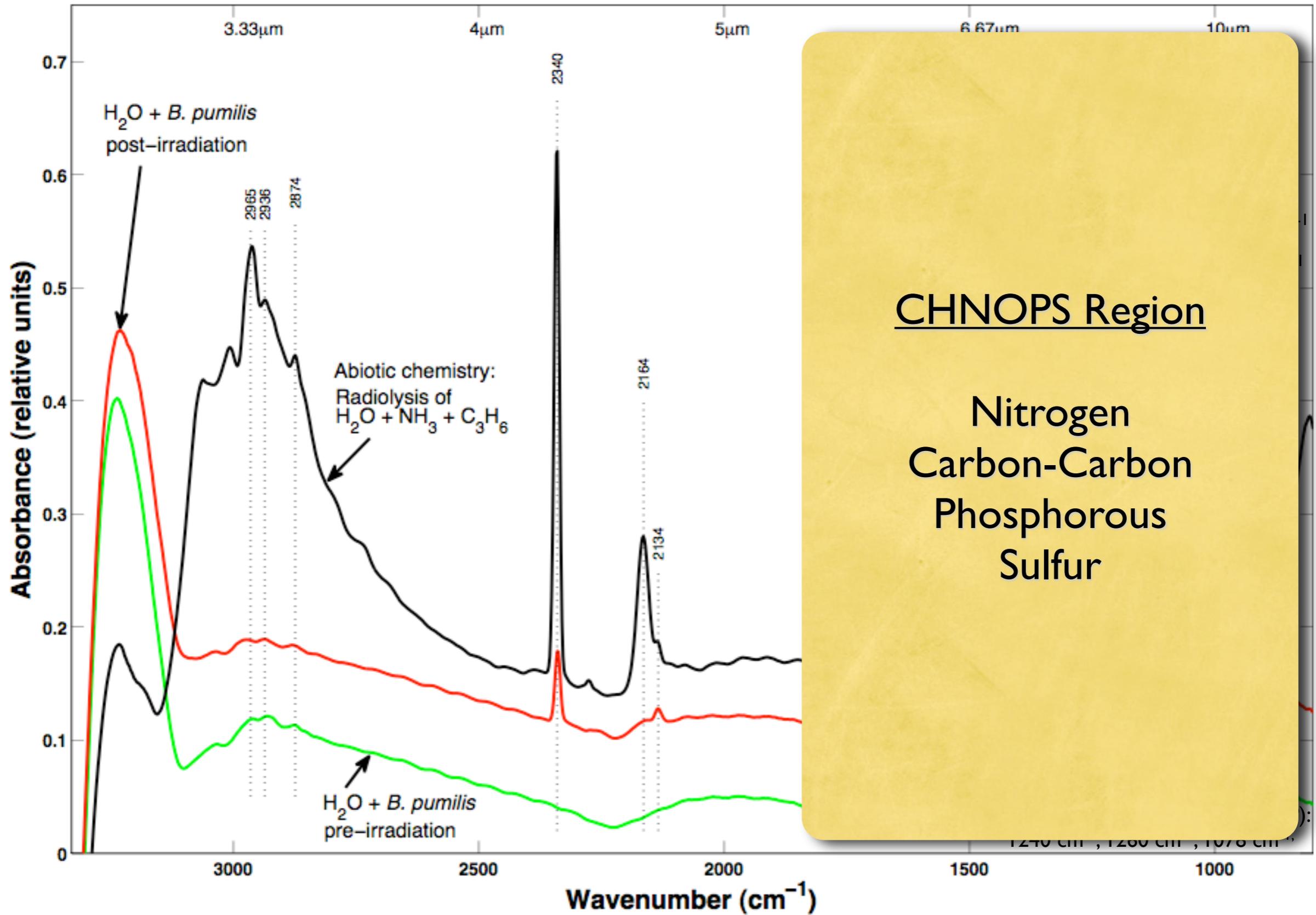
# Life Detection on Icy Worlds

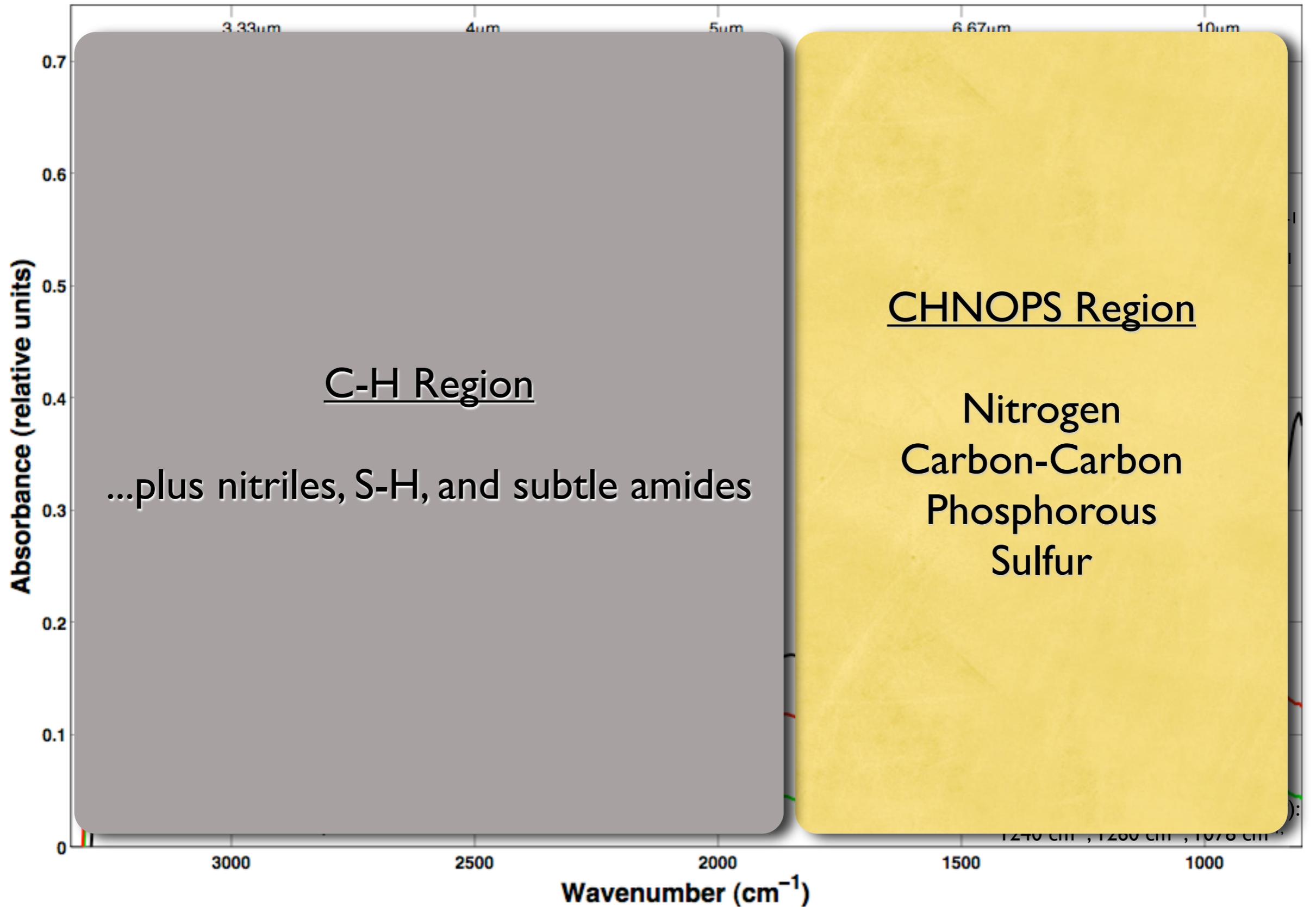
## Biosignatures vs. Abiotic Radiolytic Chemistry











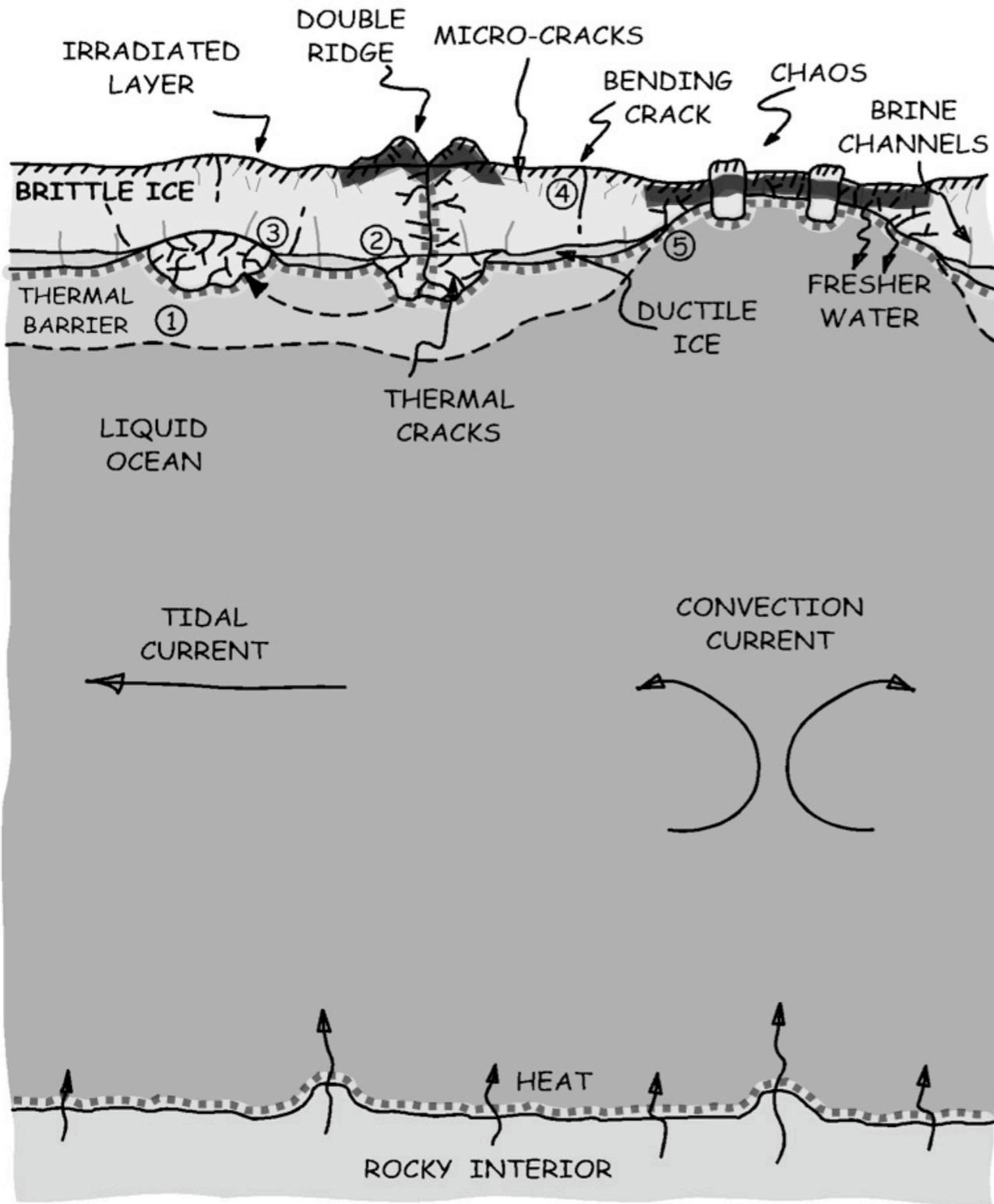
# Notional Payload Possibilities

Instrument	Source or Heritage	Mass (kg)
Accelerometer	TEAMX 2005	0.05
GCMS	TEAMX 2005	3.4
Camera	Mars Surveyor	0.35
Temp Sensor	TEAMX 2005	0.28
Raman Spectrometer	TEAMX 2005	1.5
Ion Specific Wet Chemistry Array	TEAMX 2005	0.11
Radiation Sensor	TEAMX 2005	0.005
Magnetometer	TEAMX 2005	0.7
Microseismometer	TEAMX 2005	0.3
Sample Acquisition System	TEAMX 2005	4
INMS	Huygens	9.25

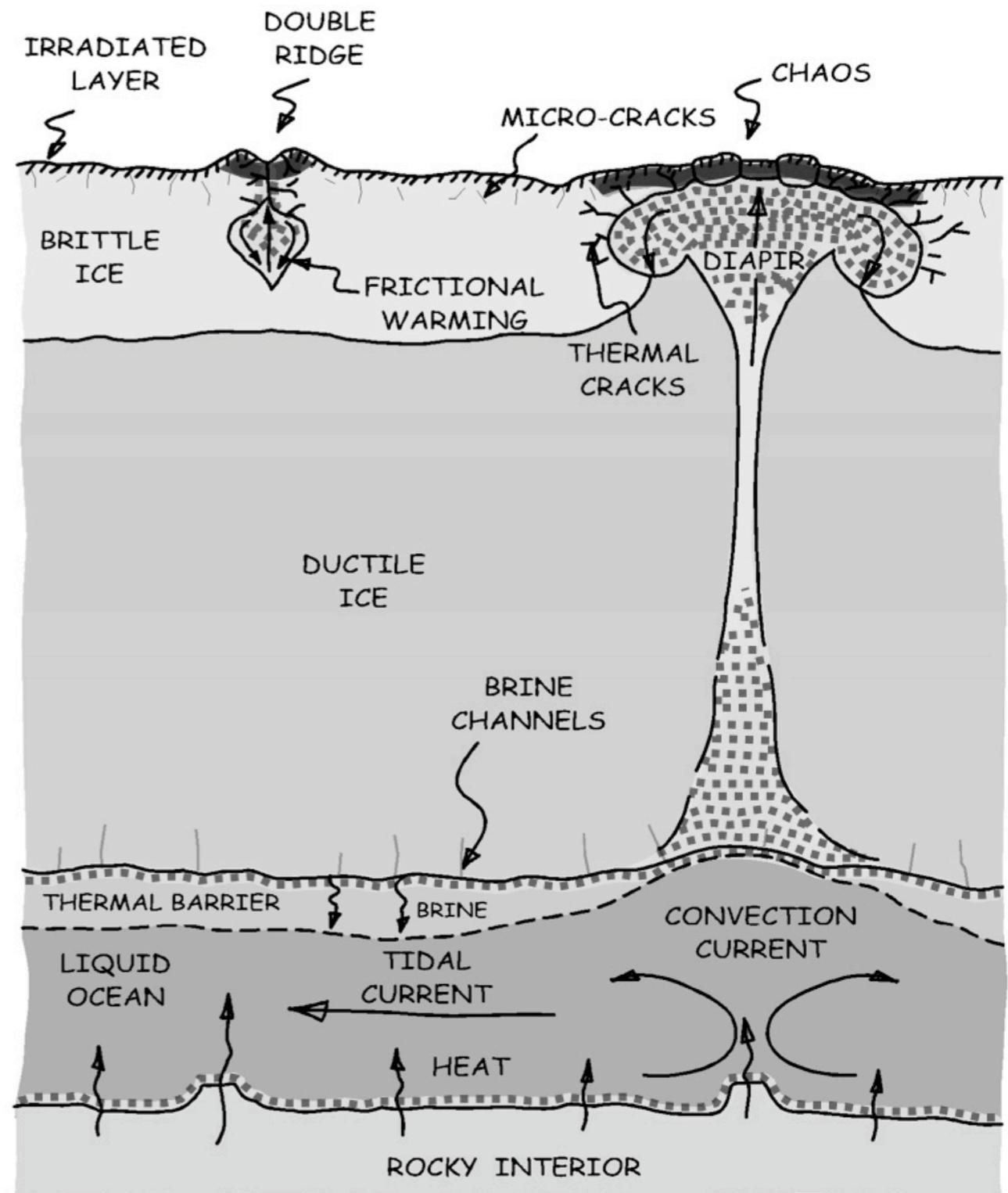
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# THIN-SHELL MODEL



# THICK-SHELL MODEL

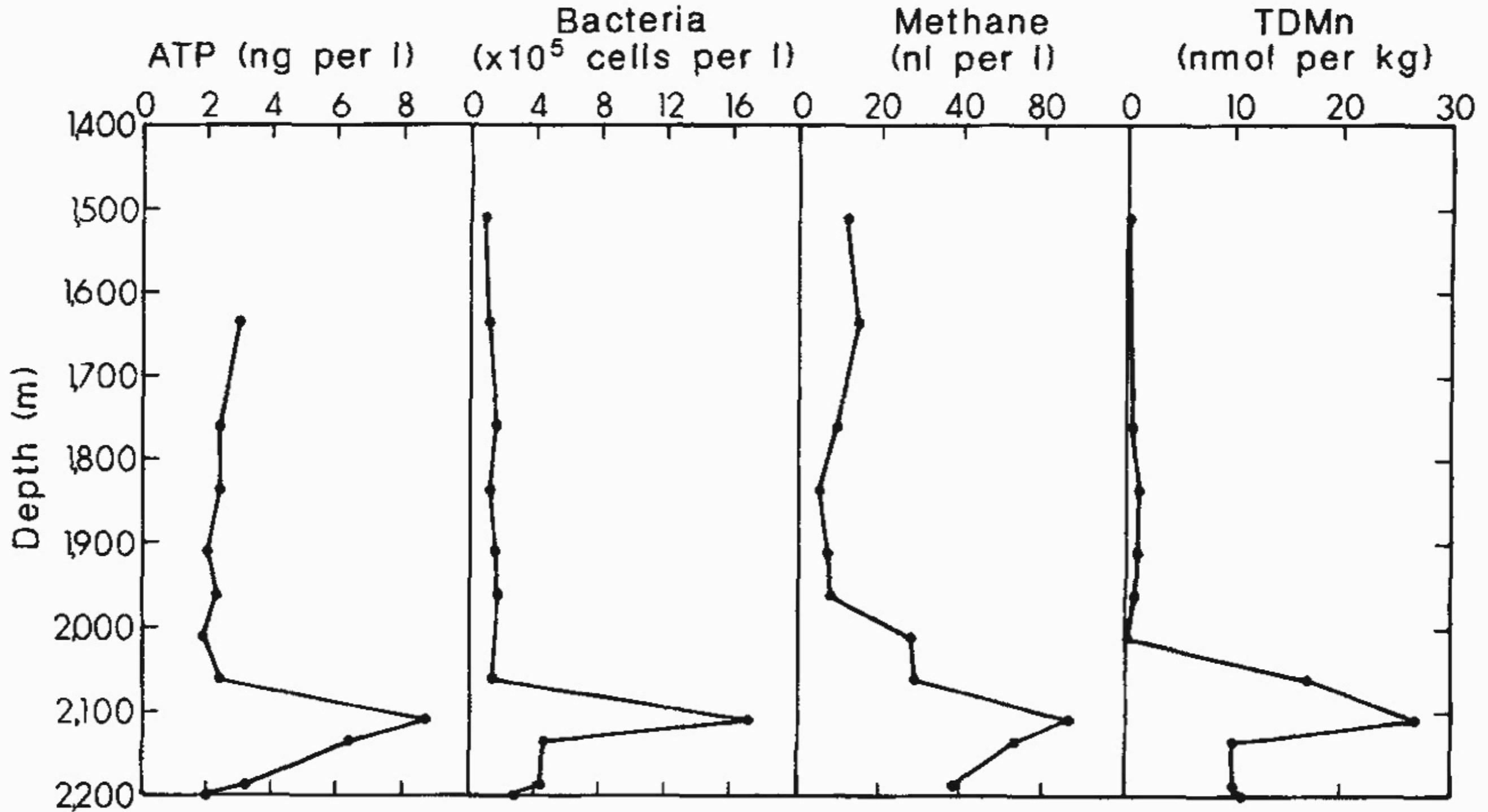


Potential Habitable zone
  Potential Biosignature Location
  Thermal Cracks
  Irradiated Layer

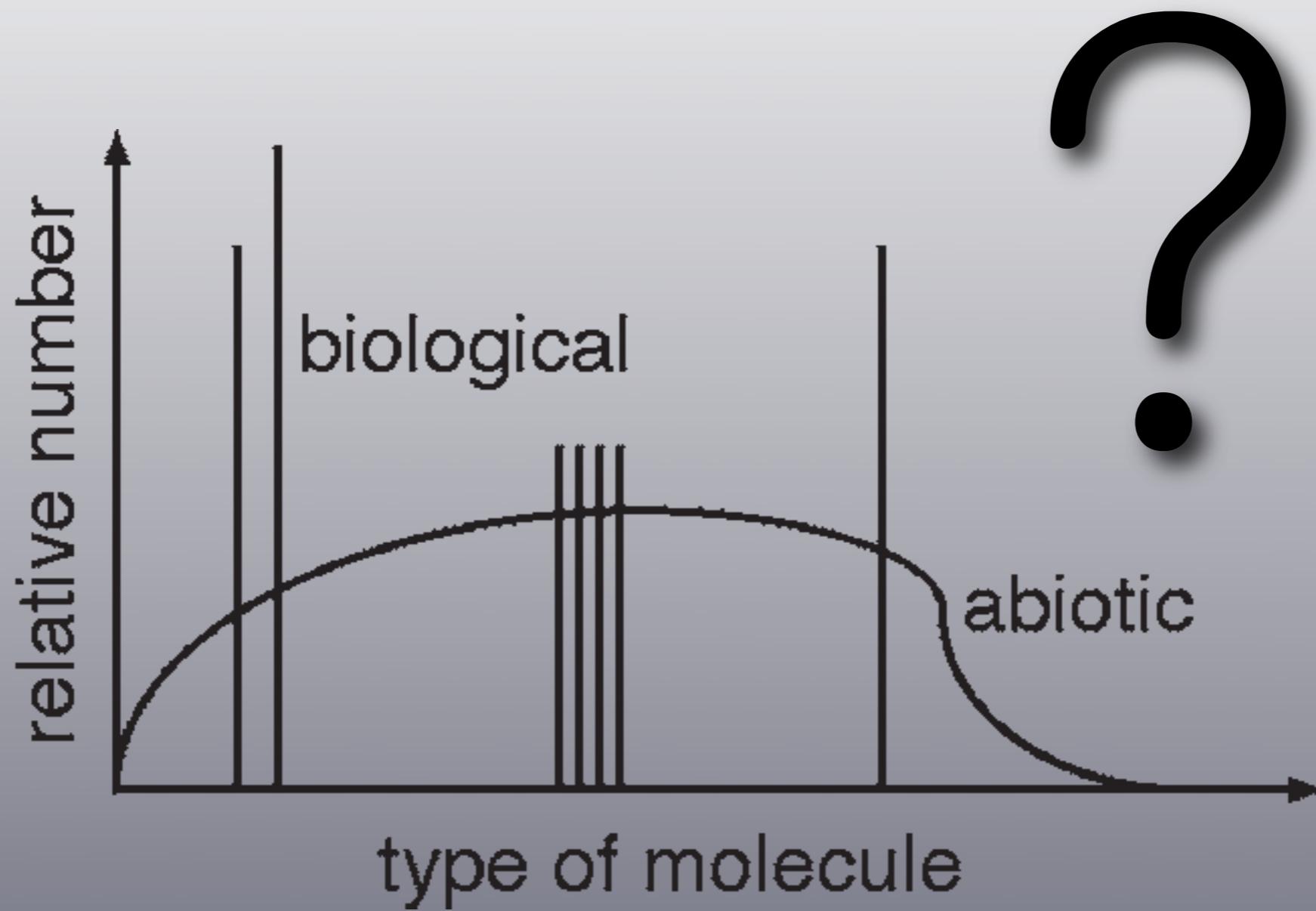
# Bacterial abundance in select terrestrial ecosystems

	Abundance (cells ml <sup>-1</sup> )	References
Ocean, surface	$5 \times 10^3 - 5 \times 10^5$	[1, 2]
Ocean, deep basins	$10^3 - 10^4$	[3]
Hydrothermal vents	$10^5 - 10^9$ (in suspension)	[3]
Hot Springs	$\sim 10^6$	[4]
Microbial mats	$10^7 - 10^9$	[6]
Sierra Snowpack	$10^3 - 10^4$	[5]
Glacial ice	120	[1]
Vostok accretion ice	83-260	[1]
Vostok water (predicted)	150	[1]

# Plumes above hydrothermal vents on Earth

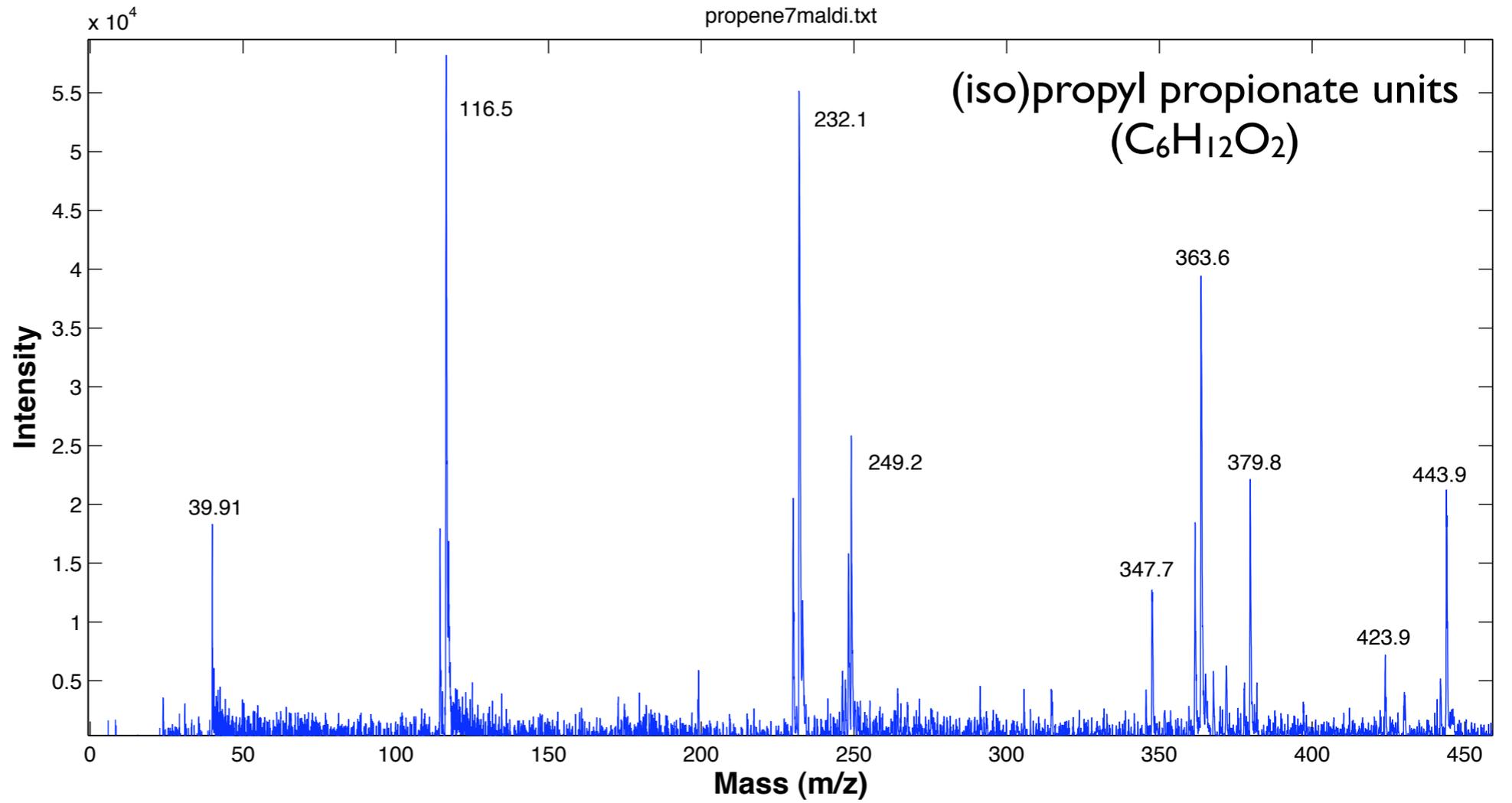


Winn and Karl (1986)

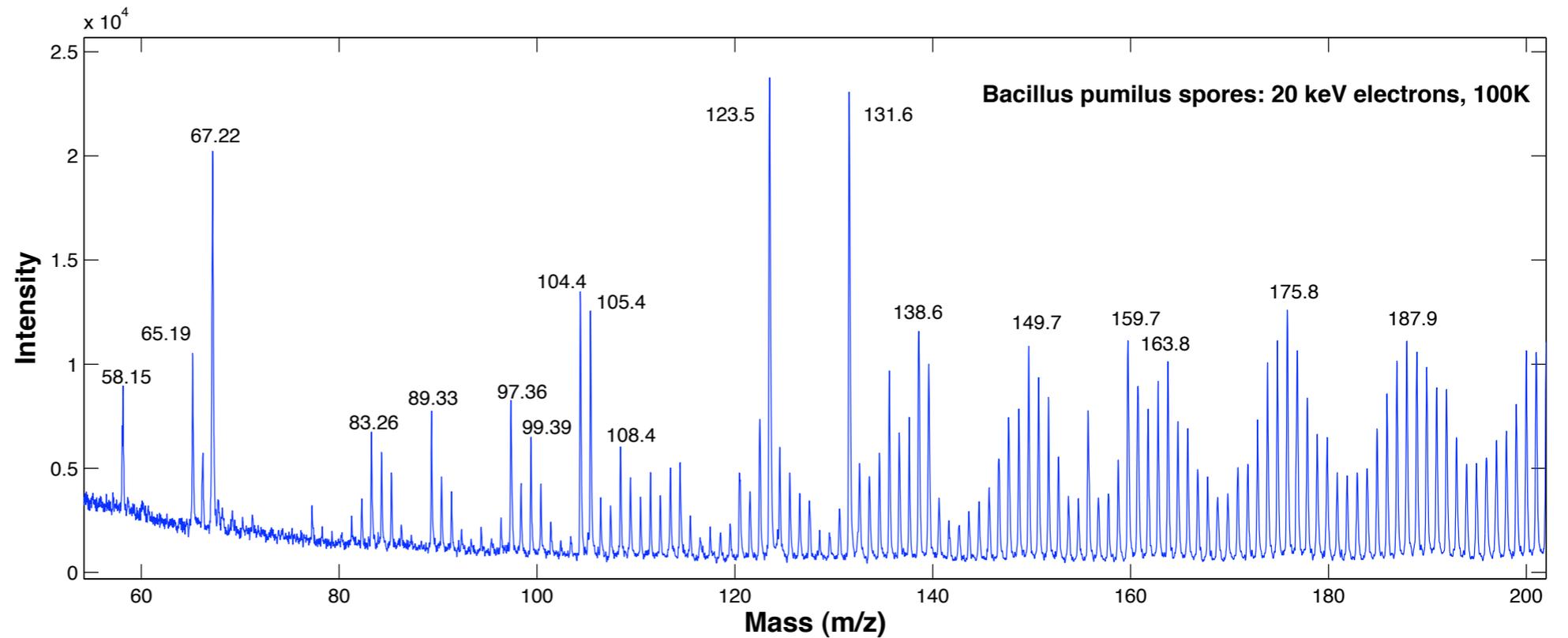


# MALDI (Matrix-Assisted Laser Desorption/Ionization) analysis:

Propene



*Bacillus pumilis*



A 3D rendered landscape of a rocky, cratered planet. The terrain is rugged with various shades of grey, blue, and brown. In the upper left, a large, yellowish, cratered moon hangs in the dark, starry sky. In the lower right, a small, metallic satellite with a gold-colored interior is positioned on the ground.

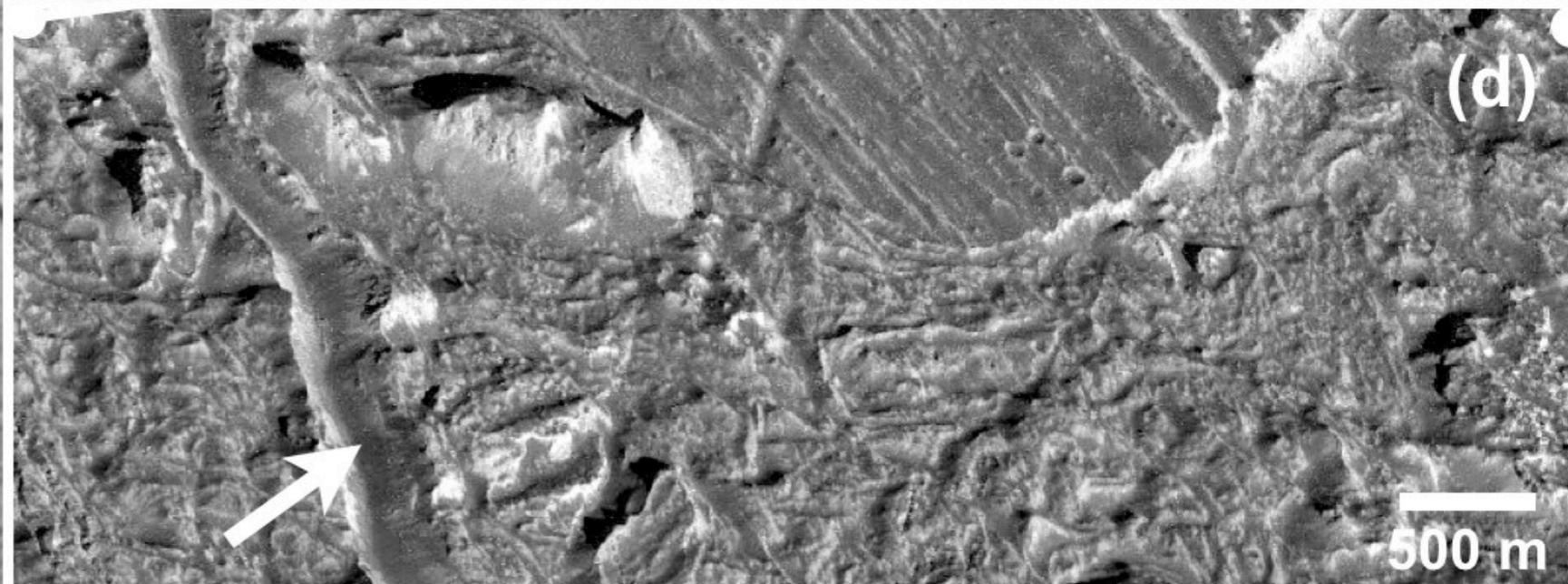
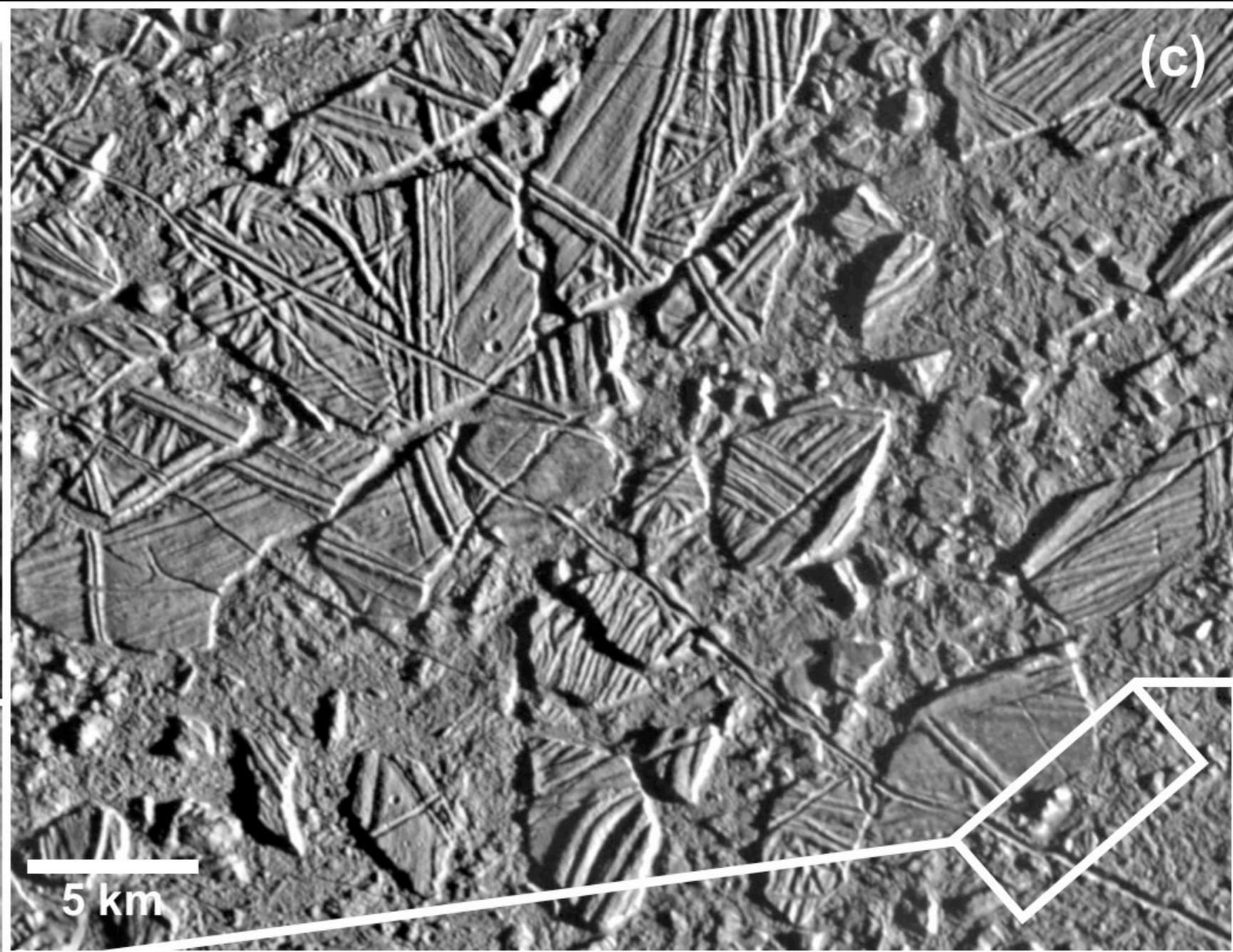
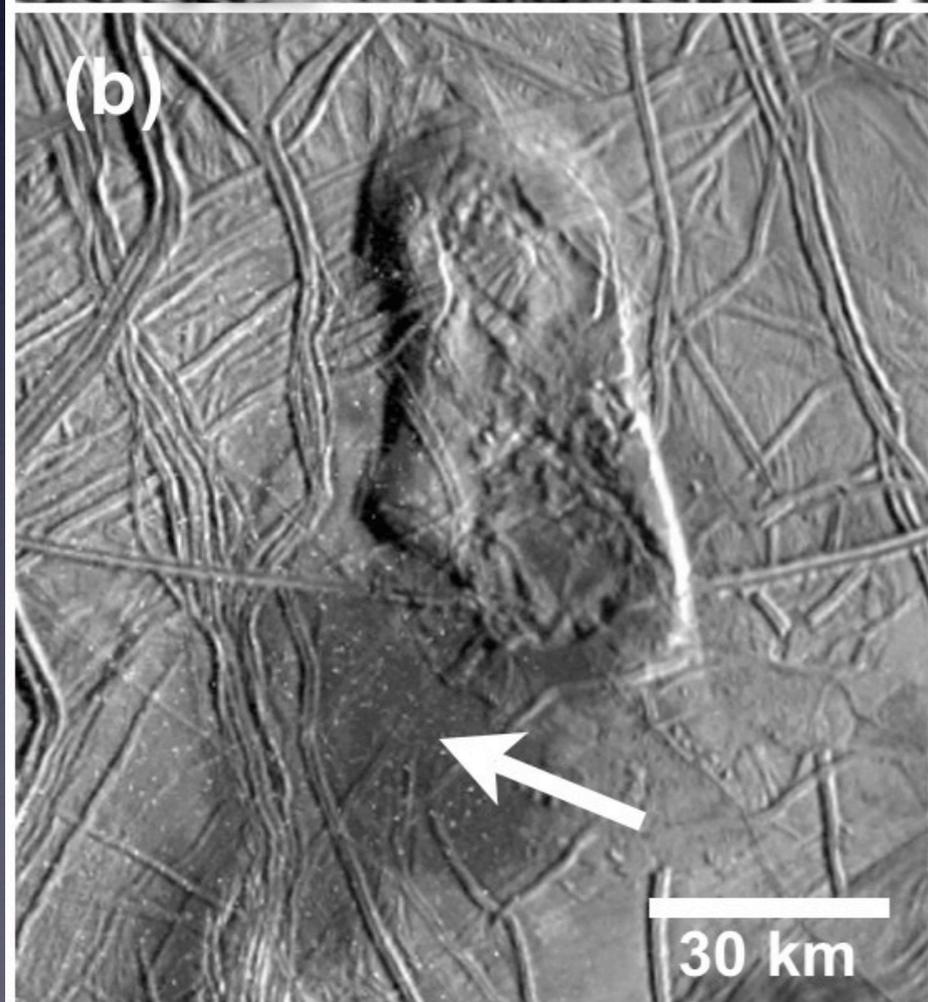
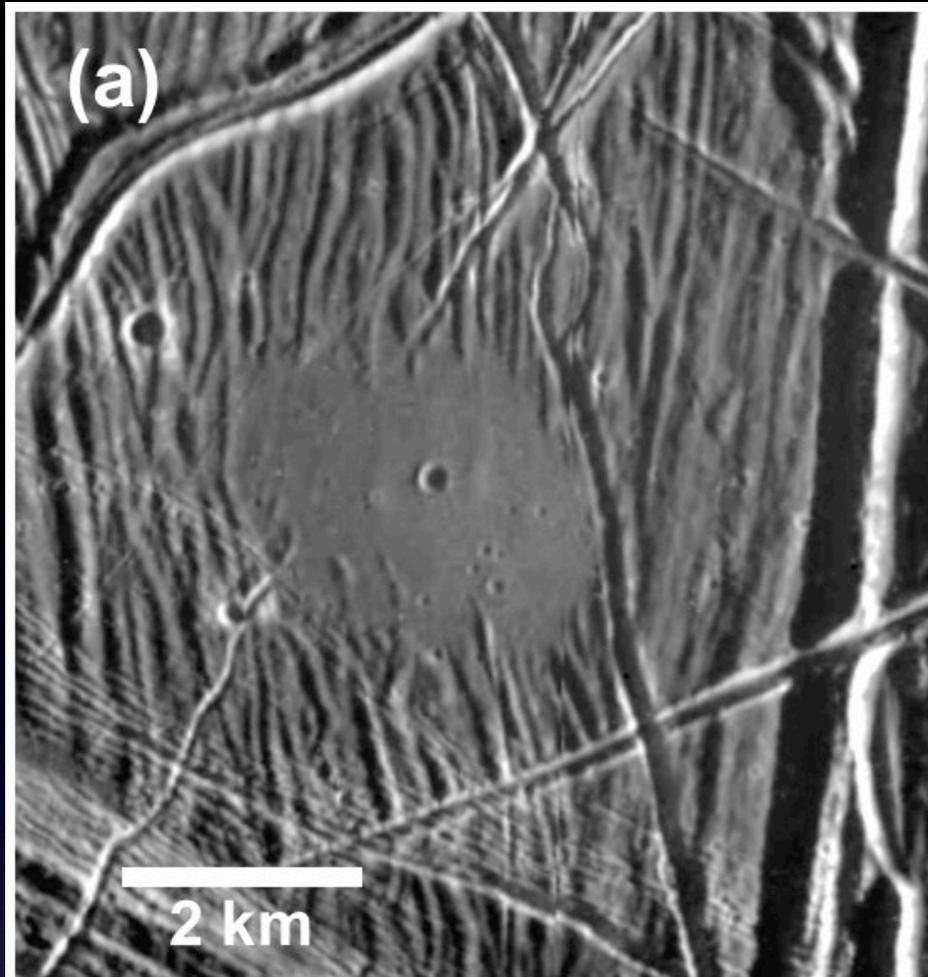
## **Acknowledgements**

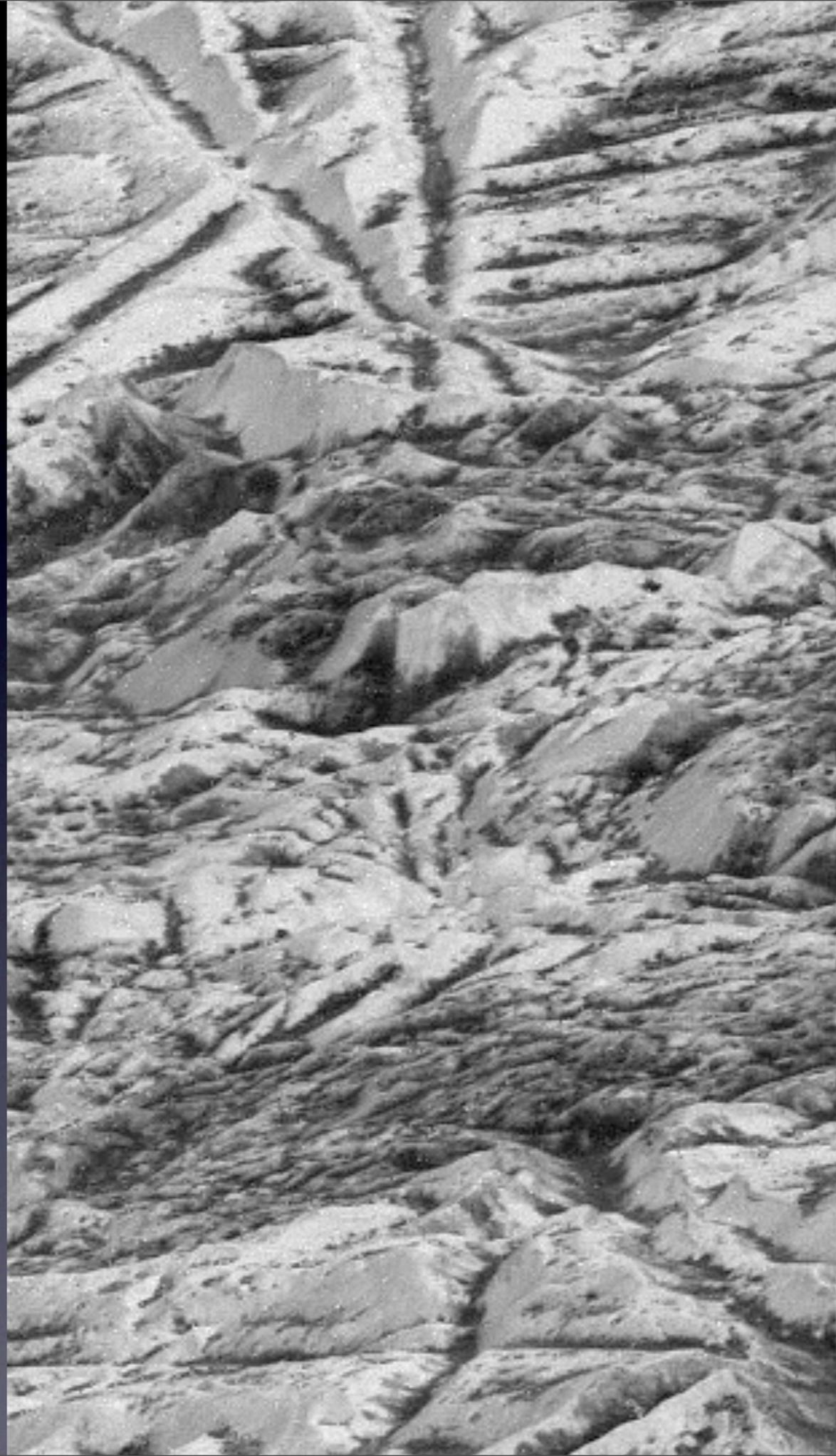
JPL-Caltech

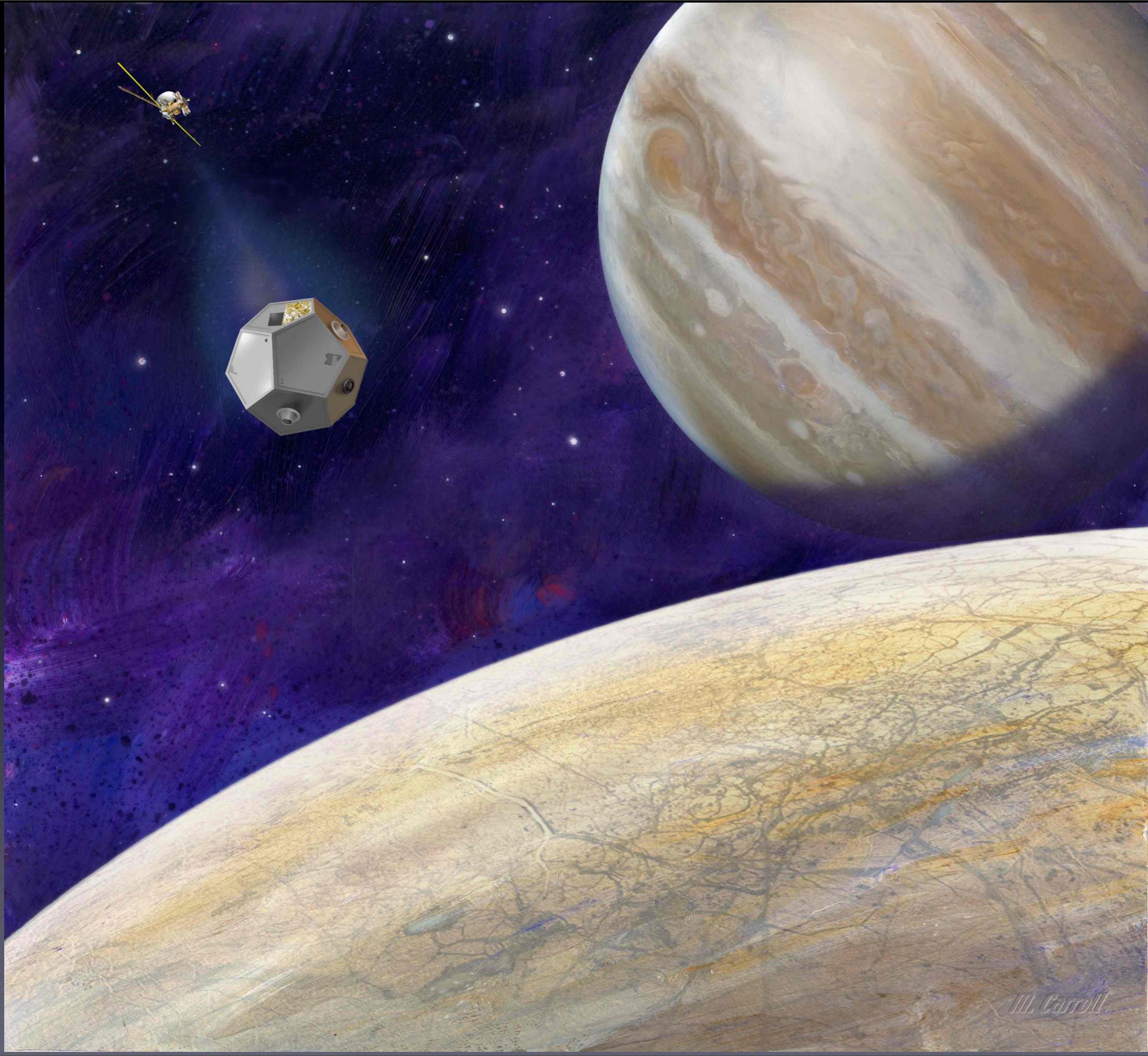
IKI, esp Anna & Andre

M Carroll (artwork)

R.W. Carlson

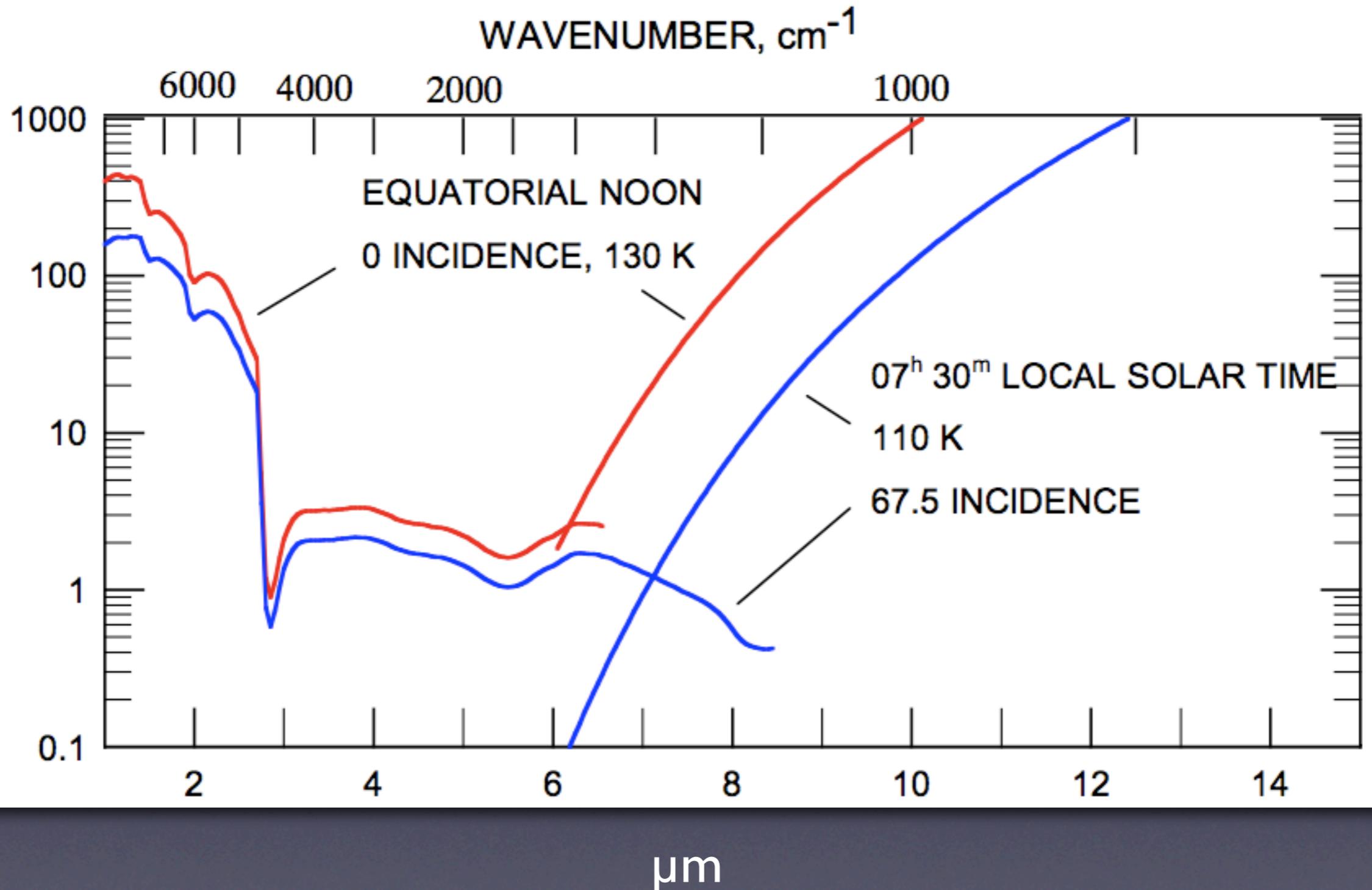






M. Carroll

Radiance,  $10^9 \text{ ph}/(\text{s}\cdot\text{cm}^2\cdot\text{sterad}\cdot\text{cm}^{-1})$



Europa surface temperature: 70-130K



# Geological criteria to guide the search for biosignatures

Figueredo et al. (2003)

- 1) Evidence for high material mobility.
- 2) Concentration of non-ice components.
- 3) Relative youth.
- 4) Textural roughness.
- 5) Evidence for stable or gradually changing environments.